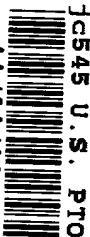


11/24/98



JCS45 U.S. PTO

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PATENT APPLICATION  
TRANSMITTAL**

(Only for new nonprovisional applications under 37 C.F.R. § 1.53(b))

Attorney Docket No.	1633-012B
First Inventor or Application Identifier	M. ADDIS
Title	Improved Brush-Seal Designs for Elastic Fluid Turbine
Express Mail Label No.	EL086798956

**APPLICATION ELEMENTS**  
See MPEP chapter 600 concerning utility patent application contents.**ADDRESS TO:** Assistant Commissioner for Patents  
Box Patent Application  
Washington, DC 20231

1. ☐ \* Fee Transmittal Form (e.g., PTO/SB/17)  
(Submit an original and a duplicate for fee processing)
  2. ☒ Specification [Total Pages (preferred arrangement set forth below)
    - Descriptive title of the invention
    - Cross References to Related Applications
    - Statement Regarding Fed sponsored R & D
    - Reference to Microfiche Appendix
    - Background of the invention
    - Brief Summary of the invention
    - Brief Description of the Drawings (if filed)
    - Detailed Description
    - Claim(s)
    - Abstract of the Disclosure
  3. ☒ Drawing(s) (35 U.S.C. 113) [Total Sheets   - 4. Oath or Declaration [Total Pages   - a. ☐ Newly executed (original or copy)
  - b. ☒ Copy from a prior application (37 C.F.R. § 1.63(d))  
(for continuation/divisional with Box 17 completed)  
(Note Box 5 below)
    - i. ☐ **DELETION OF INVENTOR(S)**  
Signed statement attached deleting inventor(s) named in the prior application, see 37 C.F.R. §§ 1.63(d)(2) and 1.33(b).
5. ☒ Incorporation By Reference (useable if Box 4b is checked)  
The entire disclosure of the prior application, from which a copy of the oath or declaration is supplied under Box 4b, is considered to be part of the disclosure of the accompanying application and is hereby incorporated by reference therein.

6. ☐ Microfiche Computer Program (Appendix)
7. Nucleotide and/or Amino Acid Sequence Submission (if applicable, all necessary)
  - a. ☐ Computer Readable Copy
  - b. ☐ Paper Copy (identical to computer copy)
  - c. ☐ Statement verifying identity of above copies

**ACCOMPANYING APPLICATION PARTS**

8. ☐ Assignment Papers (cover sheet & document(s))
9. ☐ 37 C.F.R. § 3.73(b) Statement (when there is an assignee) ☐ Power of Attorney
10. ☐ English Translation Document (if applicable)
11. ☐ Information Disclosure Statement (IDS)/PTO-1449 ☐ Copies of IDS Citations
12. ☒ Preliminary Amendment
13. ☒ Return Receipt Postcard (MPEP 503)  
(Should be specifically itemized)
14. ☐ \* Small Entity Statement(s) ☐ Statement filed in prior application, Status still proper and desired (PTO/SB/09-12)
15. ☐ Certified Copy of Priority Document(s) (if foreign priority is claimed)
16. ☐ Other: .....

**\* NOTE FOR ITEMS 1 & 14: IN ORDER TO BE ENTITLED TO PAY SMALL ENTITY FEES, A SMALL ENTITY STATEMENT IS REQUIRED (37 C.F.R. § 1.27), EXCEPT IF ONE FILED IN A PRIOR APPLICATION IS RELIED UPON (37 C.F.R. § 1.28).****17. If a CONTINUING APPLICATION, check appropriate box, and supply the requisite information below and in a preliminary amendment:**

<input type="checkbox"/> Continuation	<input checked="" type="checkbox"/> Divisional	<input type="checkbox"/> Continuation-in-part (CIP)	of prior application No: <u>08 / 891,526</u>
Prior application information: Examiner <u>n/a</u>			Group / Art Unit: <u>3108</u>

**18. CORRESPONDENCE ADDRESS**☐ Customer Number or Bar Code Label or ☒ Correspondence address below  
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Name (Print/Type)	Bradley N. Ruben	Registration No. (Attorney/Agent)	32,058
Signature		Date	24 Nov. 1998

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IN THE UNITED STATES PATENT & TRADEMARK OFFICE

Div. of Ser. No.:	Filed:	Inventor(s):	Atty Dkt:
08/891,526	14 July 1997	M. ADDIS <i>et al.</i>	1633-012B
Title: IMPROVED BRUSH-SEAL DESIGNS FOR ELASTIC FLUID TURBINES			

Box Patent Application  
Asst. Comm'r for Patents  
Washington, D.C. 20231-0001

Preliminary Amendment

Dear Sir:

In connection with the divisional application filed herewith, please first amend the application as follows:

IN THE SPECIFICATION:

Page 5, line 32, change "seals" to --seal--.

Page 9, line 3, change "form" to --forms--.

Page 15, line 21, change "portions" to --portion--; and

line 29, delete the quotation marks before "and".

Page 16, lines 18 and 27, in each change "Fig. A" to --Fig. 2A--.

Page 22, line 20, change "b" to --be--;

line 21, change "brasing" to --biasing--; and

lines 23 and 27, change "Fig. 8D" to --Fig. 8C--.

Page 26, line 22, change "Fig. B" to --Fig. 13--.

Page 34, line 30, delete "20 through 20A2" and replace with --20, 20A, and 20B--.

Page 35, line 21, change "20A1 and 20A2" to --20B and 21A separately--.

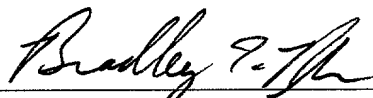
REMARKS

Entry of the foregoing amendments prior to an Examination of the subject application is respectfully requested.

These amendments correct obvious clerical and typographical errors in the application, the specification of which is a copy of the parent application. No new matter is presented.

In view of the foregoing, withdrawal of all of the rejections, and further and favorable action, in the form of a Notice of Allowance, are believed to be next in order, and such actions are earnestly solicited.

Respectfully submitted,



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24 November 1998

IMPROVED BRUSH-SEAL DESIGNS FOR ELASTIC FLUID TURBINES

Inventors:

Mark E. Addis  
William S. Dalton

5

BACKGROUND OF INVENTIONField of Invention

The present invention relates generally to improved brush seal designs for use in axially flow elastic fluid turbines.

10

Brief Description Of The Prior Art

15

The use of axial flow elastic fluid turbines, such as axial flow steam turbines, plays a very important role in the production of electric power in our society. In order produce electrical power from an electrical power generator installed at a power plant, it is necessary to rotate the rotor shaft thereof in a magnetic field produced by the stator field windings of the power generator. Typically, the torque required to rotate the rotor shaft at a sufficient angular velocity is provided by a steam turbine whose output shaft is mechanically coupled to the rotor shaft of the generator. Often, in a typical power plant, there will be a number of steam turbines each driving one or more electrical power generators.

20

25

30

In general, each steam turbine comprises a shaft rotatably supported by bearings which are encased in a housing or casting typically referred to as the turbine shell. In order to rotate the rotor shaft using the momentum of super-heated vapor (i.e., "steam"), a series of turbine stages are sequentially arranged along the axis of the shaft. A boiler, typically located external to the turbine casting, is provided for the purpose of

generating steam. External to the turbine casting are steam pipes which are used to conduct the steam from the boiler to particular sections of the turbine, that are typically classified by operating pressure. Along each section of the turbine, there are typically a number of turbine stages.

At each turbine stage, a turbine rotor is fixedly mounted to the rotor shaft. Each turbine rotor has a plurality of blades which radially extend a predetermined distance from the shaft, towards a circumferentially extending shroud band (i.e., cover) that is secured to the tenon portions of the blades. In general, each turbine blade is oriented at an acute angle with respect to the axis of rotation of its rotor. In order that each turbine rotor is permitted to freely rotate with the turbine shaft, the turbine casting has circumferential recesses to accommodate the rotor structures along the shaft. A stationary diaphragm is installed behind each rotor in a circumferential joint formed in the turbine casting. Each turbine has a ring of steam nozzles circumferentially extending about the inner structure of the diaphragm. These nozzles are located at the same radial position as the blades in its associated rotor. The function of each nozzle is to receive steam from the passageways in the turbine casting and to physically direct the steam against the rotating blades of its associated rotor. To establish a "tip seal" with the shroud band of each turbine rotor, a ring of spillstrips segments are supported from the diaphragm at each stage.

As the steam travels through the turbine, a portion of its linear momentum is transformed into the angular momentum of the rotor blades at each turbine stage, thereby imparting torque to the turbine shaft (i.e., rotor shaft). At each subsequent stage, the pressure of the steam path is typically reduced. Thus at these downstream stages it is often necessary to

increase the length of the rotor blades and the size of the associated diaphragms in order to extract kinetic energy from axially flowing steam of reduced pressure.

5 A major problem in turbine design relates to the quality of steam seals between the various stationary and rotating components along the steam flow path in the turbine. In general, there are several locations within the turbine where such seals must be established to ensure high turbine efficiency.

10 The first location where steam seals are required is between the outer portion of each rotor and its associated diaphragm have been effected using a segmented spillstrip ring of the type disclosed in U.S. Patent No. 5,547,340, incorporated herein by reference. While this Patent discloses  
15 an spillstrip design that offers improvements over prior art designs, it is not without its shortcomings and drawbacks. In particular, during start-up operations when the rotor exhibits low frequency modes of operation about its axis, the tips of the rigid fin-like structure (e.g., fin seals) projecting along  
20 the spillstrip segments tend to rub against and/or cut into the shroud-band of the associated rotor, causing damage thereto during the start-up process. The only safeguards offered against such rubbing action has been to design the spillstrips so that there exists sufficient clearance between the tip  
25 portions of the fins on the spillstrips and the shroud band of the rotor. This approach, however, results in a degrading of the tip seal, allowing steam to pass through the clearance area, and not through and over the blades of the rotor, thereby reducing the performance of the turbine.

30 The second location where steam seals are required is between the rotor and the turbine shaft. Creating seals over such regions has been addressed along time ago by installing a

segmented packing ring structure between the rotor and the turbine shaft at each turbine stage. Each packing ring structure consists of a first ring structure with multiple rows of fins (i.e., seal teeth) and a second ring structure with multiple rows of surface projections. The first ring structure is mounted from the associated diaphragm, whereas the second ring structure is mounted to the turbine shaft. Together, the rows of fins and projection structures create a labyrinth-type seal which presents a high impedance flow path to pressurized steam. However, during start up operation, low frequency modes of operation about the turbine axis tends to cause the tip portions of each row of fins to move radially outwards and inwards. To avoid tip rubbing and damage to such packing ring structures, it is necessary to design the fins and surface projections with sufficient clearance to avoid tip rubbing during start-up operation. This, however, necessarily degrades the quality of the labyrinth seal.

In U.S. Patent Nos. 4,436,311 and 5,395,124 to Brandon, the problem of fin tip rubbing in packing ring design has been addressed by providing a retractable segmented packing ring structure between each rotor and turbine shaft. The manner in which the quality of the labyrinth seal is improved with this design is described as follows. During startup operation, when low frequency rotor vibration is predominant, the diaphragm-mounted packing ring segments are spring-biased in a radial direction away from the turbine shaft, reducing the risk of fin-tip portion rubbing and packing ring damage. As the rotor increases its angular speed, low frequency vibration is naturally reduced, and the packing ring segments are forced to move closer to the turbine shaft by steam pressure, improving the quality of the labyrinth seal among the rows of fins and surface projections in the packing ring structure, thereby

improving the efficiency of the turbine.

An alternative solution to the problem of fin tip rubbing in packing ring designs is disclosed in UK Patent Application Publication No. GB 2 301 635 A. In this UK Patent Publication, a brush-type element is installed between a pair of fins extending from the packing ring segments mounted on the diaphragm. The function of the brush seal is to improve the quality of the labyrinth seal during all phases of operation. A major shortcoming with this design, however, is that during startup operations it does not provide a way of protecting the tips portions of the fin seals without designing a high degree of clearance into the design. Consequently, by virtue of such increased clearance requirements, the quality of the labyrinth seal provided by this prior art packing seal design is necessarily compromised. In addition, the brush seal design disclosed in this prior art publication is very expensive to manufacture, decreasing the price-to-performance ratio of this packing ring structure.

Thus, there is great need in the art for an improved brush seal design for providing high quality tip seals and labyrinth seals in axial flow elastic fluid turbines, while avoiding the shortcomings and drawbacks of prior art technologies.

#### OBJECTS AND SUMMARY OF THE INVENTION

Accordingly, it is a primary object of the present invention to provide an improved brush-seal design for use in creating high quality tip seals and labyrinth seals in axial flow elastic turbines, while avoiding the shortcomings and drawbacks of prior art brush-seal designs.

A further object of the present invention is to provide such a brush-seal design, which effectively reduces turbine part wear while providing an improved seals between each stage



of the turbine by accommodating for transient radial deflections of the turbine rotor and shaft during startup operation.

5 A further object of the present invention is to provide such an improved segmented spillstrip ring comprising a plurality of spillstrip segments, each containing a plurality of brush segments, wherein the angle of the end edges thereof is selected so that a different number of the same size brush segments of the present invention can be configured together to  
10 form brush-type tip seals for turbine rotors of different diameters.

A further object of the present invention is to provide such an improved segmented spillstrip ring, wherein the end surfaces of the brush segments, like the bristles thereof, extend in the direction of rotation of the turbine rotor during  
15 operation, and as the pressure across the spillstrip ring increases, the bristles of the brush segments are forced.

A further object of the present invention is to provide such an improved segmented spillstrip ring, wherein each brush segment is constructed by binding bristles between a pair of  
20 metal plates to form a brush seal strip that is installed within a groove in a tip seal segment.

A further object of the present invention is to provide such an improved segmented spillstrip ring, wherein each brush segment is constructed by directly binding bristles within a  
25 groove formed in a tip seal segment, from which a pair of spaced-apart bristle stiffening projections extend.

A further object of the present invention is to provide such an improved segmented spillstrip ring, wherein the  
30 bristles of the brush segments extend generally within or parallel to the principal plane of the turbine rotor.

A further object of the present invention is to provide

such an improved segmented spillstrip ring, wherein the  
bristles in the brush segments extend generally at an acute  
angle with respect to the principal plane of the turbine rotor,  
so that as the pressure across the spillstrip ring increases,  
5 the flexible bristles of the brush pack are forced in toward  
the rotor, thereby effectively reducing or eliminating the  
radial clearancē between the bristle and the shroud band of the  
rotor allowing for rotor misalignment and rotor excursion  
without cutting or grooving the rotor and reducing fluid flow  
10 across the tip seal.

Another object of the present invention is to provide such  
an improved segmented spillstrip ring design which can  
compensate for undersize and oversize rotors as well as rotor  
which are not concentric to the spillstrip holder.

15 A further object of the present invention is to provide an  
improved segmented packing ring for use in an elastic turbine,  
wherein an improved seal is provided between the packing ring  
holder and rotor thereof.

20 A further object of the present invention is to provide an  
improved retractable segmented packing ring for use in an  
elastic turbine, wherein an improved labyrinth seal is formed  
between the retractable segmented packing ring and the turbine  
rotor by way of multiple rows of fins and at least a single row  
of bristle elements disposed therebetween.

25 A further object of the present invention is to provide  
such an improved retractable segmented packing ring, wherein  
the bristles in the brush segments extend at an angle in the  
direction of rotation of the rotor principal plane of the  
rotor.

30 A further object of the present invention is to provide  
such an improved retractable segmented packing ring, wherein  
the bristles in the brush segments extend generally at an acute

angle with respect to the principal plane of the turbine rotor.

A further object of the present invention is to provide such an improved retractable segmented packing ring, wherein the bristles in the brush segments extend within a plane substantially parallel to the principal plane of the turbine rotor.

A further object of the present invention is to provide such an improved retractable segmented packing ring, wherein the bristles in the brush segments extend generally at an acute angle with respect to the principal plane of the turbine rotor.

An even further object of the present invention is to provide an axially flow steam turbine which incorporates novel spillstrip rings and retractable packing rings constructed in accordance with the principles of the present invention.

These and other objects of the present invention will become apparent hereinafter.

#### BRIEF DESCRIPTION OF THE DRAWINGS

For a more complete understanding of the Objects of the Present Invention, the following Detailed Description of the Illustrative Embodiment should be read in conjunction with the accompanying drawings, in which:

Fig. 1 is a perspective, partially fragmented view of a multistage axial flow steam turbine of the present invention, showing the diaphragm and rotor components of each stage housed within the turbine outer casting;

Fig. 2 is a perspective, partially fragmented view of a turbine stage of the present invention, showing (1) a portion of the first illustrative embodiment of the spillstrip ring of the present invention supported by the turbine diaphragm to form a continuous tip seal with the shroud band of its

associated rotor, and also (2) a portion of the first illustrative embodiment of the retractable packing ring of the present invention supported by the turbine diaphragm form a continuous labyrinth seal with the rotor shaft;

5 Fig. 2A is a cross-sectional view of a turbine stage of the present invention, showing the first illustrative embodiment of the spillstrip ring and the manner in which it extends from a diaphragm stationarily installed in the outer turbine casting, and establishes a continuous tip seal with the  
10 shroud band of its associated rotor;

Fig. 2B is a schematic diagram illustrating that the bristles associated with the spillstrip ring of the first and second illustrative embodiments of the present invention extend in the direction of rotation of the associated rotor;

15 Fig. 3 is a cross-sectional view of a turbine stage of the present invention, showing the first and second illustrative embodiments of the spillstrip ring of the present invention and the first illustrative embodiment of the retractable packing ring of the present invention supported from the diaphragm to form a continuous tip seal and labyrinth seal, respectively,  
20 within the turbine;

Fig. 4 is a perspective view of the first illustrative embodiment of the spillstrip of the present invention, showing the T-shaped cross-sectional dimensions of its body portion, and a single row of brush segments of the present invention embodied therein;  
25

Fig. 5 is an exploded view of a first illustrative embodiment of the brush segment (i.e., strip) according to the present invention.;

30 Fig. 5A is an elevated side view of the first illustrative embodiment of the brush segment of the present invention shown in Fig. 5;

Fig. 5B is an exploded side view of the brush segment shown in Fig. 5;

Fig. 5C is a cross-sectional view of the spillstrip of the present invention;

Fig. 6 is a perspective view of the second illustrative embodiment of the spillstrip of the present invention, showing the T-shaped cross-sectional dimensions of its body portion, a plurality of rows of rigid fins extending therefrom, and a single row of brush segments of the present invention disposed therebetween;

Fig. 7 is a cross-sectional diagram of a turbine stage according to the present invention, taken along line 7-7 in Fig. 3;

Fig. 8 is a perspective, partially fragmented view of a turbine stage of the present invention, showing a portion of the first illustrative embodiment of the retractable packing ring of the present invention and the manner in which it is supported by the turbine diaphragm to form a continuous labyrinth seal with the rotor shaft;

Fig. 8A is a first perspective view of the second illustrative embodiment of the retractable packing seal segment, showing the T-shaped cross-sectional dimensions of its body portion, the plurality of rows of rigid fins extending along the length thereof, and the single row of brush segments of the present invention embodied therein;

Fig. 8B is a second perspective view of the first embodiment of the retractable packing ring segment of the present invention;

Fig. 8C is a perspective view of a non-tractible type packing ring segment employing the brush segments of the present invention;

Fig. 9 is a cross-sectional view of the retractable

segmented packing ring of the present invention installed about the rotor shaft, taken along line 9-9 of Fig. 3;

Figs. 9A and 9B are schematic diagrams showing computer numerically controlled (CNC) machinery for use in manufacturing brush segments, spillstrip segments, and packing ring segments of the present invention;

Fig. 10 is a flow chart setting forth the steps involved in manufacturing brush segments according to the first illustrated embodiment of the present invention shown in Figs. 5 through 5B, as well as spillstrip ring segments and retractable packing ring segments shown in Figs. 3, 4A, 6 and 8A, wherein the bristles of the brush segments thereof are disposed in a plane substantially parallel to the principal plane of the associated rotor, and extend at an angle in the direction of rotation of the rotor;

Fig. 11A is a table setting forth the results of a second study showing the change of radius of curvature (i.e., drop) as a function of (i) the change in the packing ring slot diameter for packing ring segments of a fixed radius of curvature (e.g., 11.5 inches) and (ii) the length of the packing ring segment;

Fig. 11B is a table setting forth the results of a first study showing the change of radius of curvature (i.e., drop) as a function of (i) the change in the packing ring slot diameter for packing ring segments of a fixed radius of curvature (e.g., 15.5 inches) and (ii) the length of the packing ring segment;

Fig. 12A is a schematic diagram showing a quarter portion of a tip seal formed about a rotor having a first diameter (e.g., 9 inches) using a first number (e.g., 10) of fixed-length brush segments (of the first illustrative embodiment) installed within a number of custom-length size spillstrip segments;

Fig. 12B is a schematic diagram showing a quarter portion

of a tip seal formed about a rotor having a second diameter (e.g., 12 inches) using a second number (e.g., 13) of fixed-length brush segments (of the first illustrative embodiment) installed within a number of custom-length size spillstrip segments;

Fig. 13 is a perspective, partially fragmented view of a turbine stage of the present invention, showing a portion of the second illustrative embodiment of the retractable packing ring and the manner in which it is supported by the turbine diaphragm and forms a continuous labyrinth seal with the rotor shaft;

Fig. 14 is a cross-sectional view of a turbine stage of the present invention, showing the third and fourth illustrative embodiments of the spillstrip ring of the present invention and the third illustrative embodiment of the retractable packing ring of the present invention supported from the diaphragm to form a continuous tip seal(s) and labyrinth seal, respectively, within the turbine;

Fig. 14A is a schematic cross-sectional diagram of the second illustrative embodiment of the brush segment of the present invention, wherein the bristles thereof are forced downwardly by steam flow during turbine operation;

Fig. 15 is a cross-sectional view of a turbine stage of the present invention, showing the third and fourth illustrative embodiments of the spillstrip ring of the present invention and the fourth illustrative embodiment of the retractable packing ring of the present invention supported from the diaphragm to form a continuous tip seal(s) and labyrinth seal, respectively, within the turbine;

Fig. 16 is a perspective exploded view of a second illustrative embodiment of the brush segment of the present invention, wherein its bristles extend at an angle with respect

to the principal plane of the rotor when installed within the turbine system;

Fig. 16A is a cross-sectional exploded diagram of the brush segment of Fig. 16, showing its bristles (brush pack) being disposed between its front and back plates;

Fig. 17 is a flow chart setting forth the sets involved in manufacturing brush segments shown in Fig. 17, as well as spillstrip ring segments and retractable packing ring segments incorporating the same;

Fig. 18 is a perspective view of a third illustrative embodiment of the brush segment of the present invention, wherein its bristles substantially perpendicular to the longitudinal axis of the segment;

Fig. 18A is a cross-sectional, exploded diagram of the brush segment of Fig. 18, showing the bristle material, channel structure and wire core used to form the brush pack thereof, for insertion into the brush slot milled into the body portion of the packing ring segment (or a spillstrip segment of the present invention);

Fig. 18B is a cross-sectional diagram showing the brush segment of Fig. 18 installed within the slot formed between the fin-like projections milled into the body portion of the spillstrip segment (or packing ring segment) of the present invention;

Fig. 19 is a flow chart setting forth the steps involved in manufacturing brush segments shown in Figs. 18 through 18B, as well as spillstrip ring segments and packing ring segments shown in Figs. 13 through 15;

Fig. 20 is perspective view of a portion of the outer casing of the turbine of Fig. 1, showing the use of the brush seal segments according to the second illustrative embodiment of the present invention in order to seal off the joint gap



formed between the top and bottom casing portions;

Fig. 20A is cross-sectional view of the brush seal segment and casing outer halves shown in Fig. 20; and

Fig. 21 is perspective view of a portion of the outer casing of the turbine of Fig. 1, showing the use of brush seal segments according to the second illustrative embodiment of the present invention in order to seal off the joint gap formed between the top and bottom casing portions.

DETAILED DESCRIPTION OF THE ILLUSTRATIVE EMBODIMENTS  
OF THE PRESENT INVENTION

In general, the axial flow steam turbine 1 of the present invention generally comprises a number of turbine sections which are conventionally classified by operating pressure. Along each turbine section, a number of stages are encased within outer turbine casting 2 as shown in Fig. 1. In the illustrative embodiment, these turbine stages are identified by 3A, 3B, 3C, and 3D in Fig. 1. In general, however, the number of stages will vary from embodiment to embodiment of the present invention. Hereinafter, like reference numerals are used to indicate like structures.

Referring to Figs. 1 through 3, the major components of the turbine system hereof will be described.

As shown in Fig. 1, each stage of the turbine has a turbine rotor 4 fixedly mounted to the rotor (i.e., turbine) shaft 5, and an associated turbine diaphragm 6 stationary mounted in a joint 7 in the outer turbine casing 2. The position of each turbine rotor is located slightly downstream from its associated diaphragm 6. As shown in Figs. 1 and 3, the downstream flow direction is indicated by a right-hand directed arrow.

As shown in Fig. 1, turbine housing 2 is constructed using

upper and lower casing portions 2A and 2B which are bolted together in a conventional manner to form a "turbine shell". As shown, the turbine casing has a circumferential chamber 8 for accommodating the geometry of each rotor structure along the shaft, and the geometry of each turbine diaphragm. Each turbine diaphragm 6 has a ring of steam nozzles circumferentially extending about the inner and outer ring portions of the diaphragm. The radial position of the steam nozzles in any particular diaphragm is at substantially the same as the radial position of the ring of turbine blades in its associated rotor. The function of each steam nozzle in a stationary diaphragm is to receive steam from the passageways in the outer turbine casting and to physically direct this steam against the rotating blades 8 of its associated rotor. Preferably, the steam is directed against the rotating turbine blades at an angle selected to maximize momentum transfer to the turbine shaft and rotor assembly..

As shown in Figs. 2 through 3, each turbine stage of the first illustrative embodiment includes a stationary diaphragm 6 comprising: an outer ring portions 6A seated in complementary joint 7 in the outer turbine casting 2; a ring of steam directing nozzles 10 supported within the outer ring portion; and an inner ring portion 6B contained within the nozzle ring. The turbine stage includes rotor structure 4 comprising: an inner ring portion 4A attached to turbine shaft 5; a plurality of blades (i.e., buckets) 11 each being fixedly attached to inner ring portion 4A; and a circumferentially extending shroud band (i.e., blade cover) 12, which is secured to the tenon of each turbine blade. When so assembled, the blades "and the shroud band 12 are generally disposed within or about a plane centrally passing through such subcomponents. Hereinafter this plane shall be referred to as the "principal plane" of the

rotor and will be used for referencing the location and extension of the bristle used to realize the brush seal of the illustrative embodiment hereof. The rotating and stationary components comprising the various stages of the turbine are enclosed within the outer turbine casting A.

In each turbine design of the present invention, adequate steam seals between each stationary diaphragm and its associated turbine rotor are provided using two distinct steam sealing mechanisms. The details of these steam sealing mechanisms will be described below.

As shown in Fig. 2, spillstrip ring 13 of the first illustrative embodiment is seated in a complementary groove 14 formed in a spillstrip holding portion 6A that is formed as an extension on the subsequently downstream turbine diaphragm. In general, a large number of these spillstrip segment 15 are assembled together along this groove in order to construct a spillstrip ring 16, as best illustrated in Fig. 7. As shown in Fig. A, the spillstrip ring is constructed of fine metallic bristles 17 which are generally disposed in a plane substantially parallel to the principal plane of the associated rotor and extend in the direction of rotation thereof. While the spillstrip ring 16 is supported from the diaphragm by way of mated-groove structure 14, formed therein, it is understood that the method of attachment of this structure may, at any particular stage of the turbine, vary from embodiment to embodiment of the present invention.

In Fig. A and Fig. 4, the first illustrative embodiment of the spillstrip segment of the present invention is shown. As illustrated therein, each spillstrip segment 15 comprises a body portion 15A having a number of substructures, namely: horizontally disposed cross piece 15A1, which extends longitudinally, and a vertical disposed body member 15A2, which

extends substantially perpendicularly from horizontal cross piece 26. As shown, the horizontally disposed cross-piece and the vertically disposed body member form the T-shaped cross section of the spillstrip, which is precisely matched for seating in complementary groove 14 formed in diaphragm 6. As shown in Fig. 4A, a groove 15B is formed along the length of vertically disposed body member 15A2, for receipt of brush segment or strip 19 according to the present invention. Each brush segment has a pack of fine metallic bristles 20 whose end tips 20A form a bristle tip surface 21 having a radius of curvature  $r_b$  that corresponds to the radius of curvature  $r_g$  of the bottom surface of groove 15B. Likewise, the radius of curvature  $r_b$  of the bristle tip surface 21 corresponds to the radius of curvature  $r_r$  of the outer surface of the rotor. Methods for manufacturing spillstrips with such radius of curvature requirements will be described in detail hereinafter.

As best shown in Figs. 5 through 5C, the packet of fine metallic bristles 20 are bound and secured together between a pair of metal plates 22 and 23, respectively. Typically, the diameter of each bristle is in the range of about .003 to .008 inches, while the width of the bristle packet will typically be in the range of 0.040 to about .060 inches. It is understood, however, that this parameter may vary from embodiment to embodiment of the present invention. As shown, back plate 23 has a narrow tapered portion 23A that projects upwardly behind the bound bristle pack 20 below the length of the projection of the bristles above the upper end 23A of the front plate 22. Typically, the length of the bristle projection above (i.e., beyond) narrow tapered portion 23A will be in the range of about 0.030 to about 0.100 inches, although this parameter may also vary from embodiment to embodiment of the present invention. In general, this length parameter should be

selected in order to achieve: (1) an adequate seal; (2) excellent leakage reduction characteristics; and (3) excellent rubbing characteristics should rubbing occur. While the perimetrical portions of the front and back plates are machined to highly planar surfaces, to facilitate joining (TIG) and welding or brazing of these plates to form a unitary segment structure, the interior portions of the inner surfaces 22B and 23B of these plates can be channeled in a surface incised manner. The function of the incised channel is to allow the bristle pack 20 to be tightly received therein prior to the welding or brazing operation. The outer surfaces 22C and 23C of the plates 22 and 23 respectively are machined in a manner to match the geometry of the groove 15 formed in the longitudinal portion of the spillstrip body portion by, for example, press-fitting the assembled brush segment into groove 15B using conventional techniques.

As shown in Fig. 5, the end edges 22D1 and 22D2, of the front plate and end edge 23D1 and 23D2 of the back plate are machined to a planar surface that extends in the direction of pack of bristles bound between plates 22 and 23. In an illustrative embodiment, where the radius of curvature of the outer surface of the rotor is about 15 inches, the angle formed between the bristles and the lower surfaces of the front and back plates 22 and 23 is about 45 degrees, but may vary from about 30 to about 55 degrees depending on the size of the associated rotor. Typically, the length of each brush segment will be significantly shorter than the length of the spillstrip segment into which the brush segments are installed. As discovered, this allows a single (fixed-length) brush segment to be used to construct brush-type spillstrip ring segments as well as packing ring segments designed for wide range of turbine rotor diameters. The method for arriving at a suitable

length for each constituent brush segment that satisfies such design and manufacturing considerations will be described in great detail hereinafter with reference to Figs. 11A through 12B.

5 In Fig. 6, the second illustrative embodiment of the spillstrip segment hereof is shown. As in the case of the first illustrative embodiment, a large number of these spillstrip elements 25 are assembled together in an edge-to-edge manner in order to construct a continuously spillstrip ring similar to the spillstrip ring 26 shown in Fig. 7. As shown in Fig. 6, each spillstrip segment 25 comprises a body portion 25A having a number of substructures: a first horizontally disposed cross-piece portion 25A1 having a longitudinal extent; a vertically disposed body member 25A2 extending substantially perpendicularly from the cross-piece portion 25A1; and a second horizontally disposed cross-piece portion 25A3 having a longitudinal extent. As shown, each spillstrip segment further includes: multiple rows of fin-like projections 25B1 to 25B5 that extend upwardly from the top surface of the second cross-piece portion 25A3; and at least one brush-type projection 26 running between a pair of fin-like projections 25B3 and 25B4, formed from a plurality of the brush segments 27 shown in Figs 5 through 5B, whose bristle elements reside in a plane parallel to the principal phase of the rotor and extend in the direction of rotation thereof as shown in Fig. 2B. The side walls of each fin-like projection 25B1 and 25B2 are tapered and converge to a tip portion 25C having a continuous edge that also extends along the longitudinal extent of the body portion 25A. In the illustrative embodiment, each fin-like projection is machined from a single piece of metal. It is understood, however, that the fin-like projections of the spillstrip can be fabricated from a different piece of metal

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than used to fabricate vertically disposed body portion thereof. Such fin-like projections can then be press-fitted into a groove formed along the top surface of second cross-piece 25A3, in a manner similar to the way the brush segments 27 hereof are installed within the spillstrip body portion. In this embodiment of the present invention, the tip surface of the fin-like projections and the tip surface of the brush segments 27 each have a radius of curvature that corresponds with the radius of curvature of the shroud band of the associated rotor. In other embodiments, multiple rows of brush segments 27 can be installed within additional grooves formed in the spillstrip body portion. In such alternative embodiments, the bristles of each such brush segment will reside in planes substantially parallel to the principal plane of the rotor and extend in the direction of rotor rotation, as illustrated in Fig. 2B.

In Figs. 2 and 8, a portion of a segmented retractable packing ring 30 of the first illustrative embodiment is shown seated in a complementary groove 31 formed in the inner ring portion 6B of a turbine diaphragm 6. The function of the retractable packing ring 30 is to form a continuous labyrinth-type seal 32 between the inner ring portion 6B of diaphragm and the outer surface of the turbine shaft 5. As best illustrated in Figs. 8A, 8B and 9, the packing ring structure 30 of the illustrative embodiment comprises six packing ring segments 33A, 33B, 33C, 33D, 33E and 33F coupled together by coil springs 34 and the like. Alternative packing ring arrangements which may be used to practice the brush seal designs of the present invention, are described in U.S. Patent Nos. 4,436,311 and 5,395,124 of Brandon, incorporated herein by reference.

In this illustrative embodiment, each packing ring segment 33 comprises a body portion a first horizontal, extending

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portion 33A1, a vertical extending portion 33A2, and a second horizontal extending portion 33A3, contributing to the segments T-shaped cross-section as shown in Fig. 8A. In addition, each packing ring segment comprises a plurality of rows of rigid fin-like projections 33B1 through 33B9 extending along the length of its body portion; and the single row 33C of brush segments 27, (shown in Figs. 5-5C) extending along the body portion of the packing ring segment. As best shown in Fig. 8C, the row of brush-segments, 33C, is realized by installing a number of premanufactured brush segments 27 within a groove 33D that has been machined into the vertically disposed body portion 33A2 of the segment, preferably between a pair of fin rows. Each brush segment 27 within the retractable packing ring has a packet of metallic bristles which reside in a plane substantially parallel to the principal plane of the associated rotor and extend in the direction of rotation of the turbine shaft. The tips of these bristles terminate along a bristle tip surface 34 having a radius of curvature  $r_b$  that corresponds to the radius of curvature  $r_f$  of the outer surface of the turbine shaft. Likewise, the radius of curvature of the rear surface 23E of the brush segment corresponds to the radius of curvature of the bottom surface 33D1 of groove 33D, shown in Figs 8A and 8C. Methods for manufacturing packing segments with such radius of curvature requirements will be described in detail hereinafter.

In order to create a net biasing force in the radially outward direction (away from the turbine shaft), adjacent packing ring segments of the illustrative embodiment, namely 33A and 33F, 33F and 33E, 33D and 33C, and 33C and 33D, are mechanically coupled by a coil-type spring 34 which is fitted into a cylindrical bore 33F formed in the side edge surface of each such packing ring segment, as shown in Figs. 8A and 9.



Collectively, these coil springs cooperate to produce an array of forces that bias the packing ring segments in a radially-outward direction. During the start-up phase of turbine operation, when the rotor has the greatest tendency to wobble about the rotational axis of the turbine shaft (e.g., undergo transient radial displacements), the rows of fin-like projections associated with the packing ring are located at their maximum distance from the corresponding surfaces associated with the turbine shaft, while the tips of the brush segments maintain contact with the surface of turbine shaft. As the speed of the turbine increases, the pressure of steam behind each packing ring segment forces the packing rings to move radially inward against the outwardly directed spring biasing forces, causing the fin-like projections of the packing ring to move closer towards the opposing surfaces on the turbine shaft, thereby increasing the quality of the labyrinth-type steam seal. Based on the expected load characteristics at each turbine stage, the spring characteristics will be selected in a manner well known in the art. In other embodiments, a different number of packing ring segments may be used, with different coupling mechanisms used to create biasing forces for segment retractability.

In Fig. 8D, an alternative embodiment of the packing ring segment of the present invention 36 is shown. This packing ring segment 36 is non-retractible and employs a flat spring 37 to secure the packing ring segment in its mounting groove 31. As shown in Fig. 8D, a groove 36A is formed in the body portion of the segment, for receipt of end portion 37A of flat spring 37. In all other respects, packing ring segment 36 is similar to packing ring segment 33, thus embodying the brush-sealing feature of the present invention.

Referring to Fig. 10, a method for manufacturing the brush

segments 27, spillstrip segments 15, 25 and packing ring segments 33 of the present invention will now be described below.

As indicated at Block A in Fig. 10, the first step of the generalized method involves premanufacturing the front and back plates 22 and 23 of each brush segment 27 as shown in Fig. 5A. These plates can be milled, molded or forged using techniques well known in the art. The size and length of these plates can be determined using the "drop study" information tables set forth in Figs. 11A and 11B. The use of these information tables will be described in detail hereinafter.

As indicated in Block B of Fig. 10, the elongated bristles of a brush pack are loaded within the incised channels formed on the interior surface of the plates. Automated machinery can be used to hold the plates 22 and 23 in place while a packet of bristles 20 are arranged between the plates. After the packet of bristles have been arranged between the front and back plates, the plates are brought together and welded, as indicated in Block C in Fig. 10. Typically, these assembled brush segments will be manufactured in several standard arch lengths that can be used to construct a wide range of spillstrip segments and packing ring segments in accordance with the principles of the present invention. At Block D, the proper size brush segments are selected for manufacturing the segmented spillstrip ring or packing ring under design.

At Blocks E through G, packing ring segments or spillstrip ring segments are manufactured by a subprocess that is separate from the subprocess specified at Blocks A through C, described above. At Block E, the design parameters for the spillstrip ring or packing ring are acquired using measurement techniques well known in the art. At each turbine stage of interest, the design parameters to be acquired for spillstrip ring designs

will typically include: the radius of curvature of the rotor; the radius of curvature of the surface of the groove in diaphragm receiving the spillstrip ring segments; etc. The design parameters to be acquired for packing ring designs will typically include: the radius of curvature of the turbine shaft and its packing ring surfaces mounted thereabout; the radius of curvature of the surface of the groove in the diaphragm receiving the packing ring segments; the pitch and number of labyrinth steps (lands) on the turbine shaft; the clearance existing between the turbine shaft and packing ring; the "as found" wear of the packing ring fins (teech); etc. Once collected, such measurements are recorded for subsequent use in manufacturing the spillstrip and packing ring segments of the present invention.

As indicated at Block F, the design parameters acquired at Block E for the spillstrip ring design are then used to manufacture the body portion of each spillstrip ring segment.

Likewise, the design parameters acquired at Block E for the packing ring design are then used to manufacture the body portion of each spillstrip ring segment. Conventional CNC milling equipment can be used to manufacture the body portions of the spillstrip segments and the packing ring segments.

As indicated at Block G in Fig. 10, a groove is milled in the inner diameter of the spillstrip segment (or the packing ring segment) in order to accept brush segments manufactured at Blocks A through C, described above. The CNC milling equipment shown in Figs. 9A and 9B can be used to machine the required radius of curvature surfaces into the spillstrip body portion. This same equipment can be easily modified for milling surfaces in the body portion packing ring segments according to the present invention. As indicated at Block H, manufactured brush segments are inserted into the groove milled into the body

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portion of the spillstrip segments (or packing segments as the case may be). This step of the manufacturing process can be carried out using robotic machinery and the like. Once inserted within the groove of the body portion of the spillstrip segment (or packing ring segment), the individual brush segments are welded into place, as indicated at Block I. The assembled spillstrip segments are then installed into a trimming fixture, with a clamp stiffener disposed about the bristles, as indicated at Block I. As indicated at Block K, the end portion of the bristles are trimmed using a cutter so that the tip portions thereof conform to a surface having a radius of curvature that corresponds with the radius of curvature of the shroud band of the rotor at the applicable turbine stage. The apparatus illustrated in Figs. 9A and 9B can be used to carry out such brush trimming operations. In the case of spillstrip segment manufacture, the radius of curvature associated with the cutting surface of the milling machine is set to the radius of curvature of the shroud band of the rotor. In the case of packing ring segment manufacture, the radius of curvature associated with the cutting surface of the milling machine is set to the radius of curvature of the corresponding surface on the turbine shaft.

In the information tables of Figs. 11A and 11B, the results of a "drop" study for segmented brush seals are presented. As indicated by the computed "drop" measures listed in these information tables, a brush segment having a segment length of 1 1/4 inches and a 11.500 inch radius of curvature, can be used to create continuous brush seals with packing ring segments having radius of curvatures within the range of 20-27 inches. Similarly, a brush segment having a segment length of 1 1/2 inches and a 15.500 inch radius of curvature, can be used to create continuous brush seals within packing ring segments

having radius of curvatures within the range of 28-35 inches. Thus, by virtue of the present invention, it is possible make and stock two size brush segments for use in realizing continuous brush-type labyrinth seals in retractable packing rings having radius of curvatures within the range of from about 20 to about 35 inches.

Similarly, a brush segment of fixed-size can be used to form continuous brush-type tip seals about turbine rotors having different size diameters. This application is illustrated in Figs. 12A and 12B, wherein a single-size brush segment of the present invention is used to create continuous brush-type tip seals about two rotors having 9 inch and 12 inch diameters. The primary difference between these two designs is that more brush segments are required to form the brush seal about the 12 inch diameter rotor than the 9 inch rotor.

This technique should allow for a substantial reduction in the cost of manufacture of both spillstrips rings and packing rings.

In Fig. 13, the second illustrative embodiment of the retractable packing ring of the present invention is shown. The primary difference between the packing ring segment 33' shown in Fig. B and the packing ring segment 33 shown in Fig. 8 is that the brush segments 39 installed within the body portion of each packing ring segment 33' are canted in the direction opposite of steam flow, that is, each bristle is disposed at an acute angle with respect to the principal plane of the associated rotor and extends substantially within the plane formed by a radius of the rotor and the axis of rotation of the rotor, as best illustrated in Figs. 13, 14 and 14A. Also, in the alternative brush seal design of Fig. 13, the bristles within each brush segment extend in a radially inward direction, rather than in the direction of rotor rotation as

shown in the case of the brush design of Figs. 5 through 5B. The advantage of this brush seal design is that when the steam pressure within the turbine stage increases, the steam pressure forces the tips of the bristles against the rotor shroud band surface, thereby increasing the steam seal. As indicated in Fig. 14, the canted brush seal design can be implemented in either a spillstrip ring structure or a (retractable or non-retractable) packing ring structure, as shown in Fig. 14A. In the turbine design shown in Fig. 14, the third embodiment of the spillstrip segment 15', mounted on the lower pressure side of the illustrated turbine stage, includes a single row of canted brush segments installed within the body portion of the spillstrip ring. The fourth embodiment of the spillstrip segment 25', mounted on the higher pressure side of the turbine stage, includes a pair of canted brush segment rows installed within the body portion of the spillstrip ring.

As shown in Fig. 14, each retractable packing ring segment of the second illustrative embodiment 33' has a single row of canted brush segments 39 disposed between a plurality of rows of rigid fins vertically extending from the body portion of each packing ring segment. The bristles in each packing ring segment 33' extend in a radially inward direction and are canted in the direction opposite of steam flow, as described above

In the third illustrative embodiment of the retractable packing ring 33", shown in Fig. 15, each row of fins along the packing ring segment is realized by a row of canted brush segments 39, as described above.

In Fig. 16, the design and construction of brush segment 39 of the present invention is illustrated. As shown in Fig. 16A, a packet of bristles (e.g., brush pack) is disposed between the interior surface front and back plates 41 and 42,

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respectively. In this embodiment of the present invention, each bristle is arranged substantially perpendicular to the bottom edge surface 39A of the assembled brush segment, as illustrated in Fig. 16. This ensures that the bristles will reside within a plane that radially extends through the associated rotor and along the axis of rotation of the shaft when installed within the body portion of the spillstrip, while extending in the opposite direction of steam flow (e.g., against the pressure gradient). The front and back plates 41 and 42 are welded together to form a unitary structure. As shown in Fig. 16A, the packet of bristles are disposed at an acute angle with respect to the bottom edge surface 41A of plate 41. This ensures that the bristles are canted with respect to the principal plane of the rotor when installed within the body portion of the spillstrip ring.

Referring to Fig. 17, a method for manufacturing the canted-type brush segments 39, spillstrip segments 15' and 25', and retractable packing ring segments 33' and 33' incorporating the same, will now be described in detail below.

As indicated at Block A in Fig. 17, the first step of the generalized method involves premanufacturing canted front and back plates 41 and 42 shown in Fig. 16A. These plates can be milled, molded or forged using techniques well known in the art.

As indicated in Block B of Fig. 17, a packet of elongated metallic bristles are arranged between the interior surface of the plates 41 and 42. Automated machinery can be used to hold the plates in place while the bristle packet 40 is arranged therebetween. When the bristles are arranged between the plates, the plates are brought together and welded, as indicated in Block C in Fig. 17. Typically, these assembled brush segments will be manufactured in several standard arc

lengths, which can be used to construct a wide range of spillstrip segments and packing ring segments according to the present invention. At Block D, the proper size brush segments are selected for manufacturing the segmented spillstrip ring or packing ring under design.

At Blocks E through G in Fig. 17, packing ring segments or spillstrip ring segments are manufactured by a subprocess independent from the subprocess specified at Blocks A through C, described above. At Block E, the design parameters for the spillstrip ring or packing ring are acquired using measurement techniques well known in the art. At each turbine stage of interest, the design parameters to be acquired for spillstrip ring designs will typically include: the radius of curvature of the rotor; the radius of curvature of the surface of the groove in the diaphragm receiving the spillstrip ring segments; the width of the rotor (wheel) shroud band; the clearance existing between the turbine shaft and spillstrip; the "as found" wear of the spillstrip fins (teeth); etc. The design parameters to be acquired for packing ring designs will typically include: the radius of curvature of the turbine shaft and its packing ring surfaces mounted thereabout; the radius of curvature of the surface of the groove in the diaphragm receiving the packing ring segments; the pitch and number of labyrinth steps (lands) on the turbine shaft; the clearance exiting between the turbine shaft and packing ring; the "as found" wear of the packing ring fins (teeth); etc. Once collected, such parameter measurements are recorded for use in manufacturing the spillstrip ring segments and packing ring segments of the present invention.

As indicated at Block F, the design parameters acquired at Block E for the spillstrip ring design are then used to manufacture the body portion of each spillstrip ring segment.



Likewise, the design parameters acquired at Block E for the packing ring design are then used to manufacture the body portion of each spillstrip ring segment. Conventional CNC milling equipment can be used to manufacture the body portions of the spillstrip segments and the packing ring segments.

As indicated at Block G in Fig. 17, a groove is milled in the inner diameter of the spillstrip segment (or the packing ring segment) in order to accept brush segments manufactured at Blocks A through C, described above. The CNC milling equipment shown in Figs. 9A and 9B can be used to machine the required radius of curvature surfaces into the spillstrip body portion. This same equipment can be easily modified for milling surfaces in the body portion packing ring segments according to the present invention. As indicated at Block H, manufactured brush (seal) segments are inserted into the groove milled into the body portion of the spillstrip segments (or packing segments as the case may be). This step of the manufacturing process can be carried out using robotic machinery and the like. Once inserted within the groove of the body portion (of the spillstrip segment or packing ring segment), the individual brush segments are welded into place, as indicated at Block I. The assembled spillstrip segments are then installed into a trimming fixture, with a clamp stiffener disposed about the bristles, as indicated at Block I. As indicated at Block K in Fig. 17, the end portion of the bristles are trimmed using a shaper cutter so that the tip portions thereof conform to a surface having a radius of curvature that corresponds with the radius of curvature of the shroudband of the rotor. The apparatus illustrated in Figs. 9A and 9B can be used to carry out such brush trimming operations. In the case of spillstrip segment manufacture, the radius of curvature associated with the cutting surface of the milling machine is set to the radius

of curvature of the shroud band of the rotor. In the case of packing ring segment manufacture, the radius of curvature associated with the cutting surface of the milling machine is set to the radius of curvature of the corresponding surface on the turbine shaft.

In Fig. 18, a third illustrative embodiment of the brush segment of the present invention, 45, is shown. In this alternative brush segment design, each bristle strip 46 of desired length is formed by taking bristle material 47 (approximately twice the desired the bristle length), folding the material about a wire core 48, and then binding the folded bristle material within a rectilinearly extending channel structure 49, as shown in Figs. 18A and 18B. Preferably, the channel structure 49 is made from metal, but other materials may be suitable in particular applications. When the brush strip 45 is completely assembled, the wire core 48 is securely held between the walls of the channel structure 49, thereby securing the bristles therewith. When assembled, the bristles extend perpendicularly from the rear surface 45A of the bound bristle strip 45. Also, when installed within a turbine system, the bristles extend at an angle with respect to the principal plane of the rotor.

As shown in Figs. 18A and 18B, the assembled brush strip 45 is inserted into a brush slot 50 milled into the body portion of a packing ring segment or into the body portion of a spillstrip segment. The groove and its surfaces can be milled in order to conform the radius of curvature requirements of the particular turbine stage of concern. In the embodiment shown in Figs. 18A and 18B, slot 50 is formed between a pair of fin-like projections 51 and 52 milled into the body portion of the spillstrip segment (or packing ring segment). The bristles extend from milled slot 50 and extend slightly beyond rear fin-

like projection 52, which is larger than projection 51, ut  
slight shorter than bristle 53.

Referring to Fig. 19, a method for manufacturing the  
canted-type spillstrip segments and packing ring segments  
incorporating brush strips 45 will now be described below.

As indicated at Block A in Fig. 19, the first step of the  
generalized method involves producing rectilinear bound brush  
strips (i.e., segments) 45 of the required length. At Blocks  
B through H in Fig. 19, packing ring segments or spillstrip  
ring segments are manufactured by a subprocess independent from  
the subprocess represented at Block A. At Block B, the design  
parameters for the spillstrip ring or packing ring are acquired  
using measurement techniques well known in the art. At each  
turbine stage of interest, the design parameters to be acquired  
for spillstrip ring designs will typically include: the radius  
of curvature of the rotor; the radius of curvature of the  
surface of the groove in diaphragm receiving the spillstrip  
ring segments; the width of the rotor (wheel) shroud band; the  
clearing existing between the turbine shaft and spillstrip; the  
"as found" wear of the spillstrip fins (teeth); etc. The  
design parameters to be acquired for packing ring designs will  
typically include: the radius of curvature of the turbine shaft  
and its packing ring surfaces mounted thereabout; the radius of  
curvature of the surface of the groove in diaphragm receiving  
the packing ring segments; the pitch and number of labyrinth  
steps (lands) on the turbine shaft; the clearing existing  
between the turbine shaft and packing ring; the "as found" wear  
of the packing ring fins (teeth); etc. Once collected, such  
parameter measurements are recorded for use in manufacturing  
the spillstrip ring segments and packing ring segments of the  
present invention.

As indicated at Block C, the design parameters acquired at

Block B are then used to manufacture (e.g., mill) the body portion of each spillstrip ring segment (or packing ring segment) with a canted groove 54, as illustrated in Fig. 15. Conventional CNC milling equipment can be used to manufacture the canted grooves in the body portions of the spillstrip segments and the packing ring segments.

As indicated at Block D in Fig. 19, the acquired designed parameters are used to bend the rectilinear bound brush strip 45 to the appropriate radius of curvature. Thereafter, as indicated at Block F, the brush strip is inserted into the canted groove in the body portion of the spillstrip ring segment or packing ring segment. As indicated at Block G, the inserted bristle strip is then (TIG) welded, brazed or staked into its canted groove in the spillstrip ring segment or packing ring segment. These steps of the manufacturing process can be carried out using robotic machinery and the like. As indicated at Block G, the assembled spillstrip segments are then installed into a trimming fixture, with a clamp stiffener disposed about the bristles. As indicated at Block H in Fig. 19, the end portion of the bristles are trimmed using a cutter so that the tip portions thereof conform to a surface having a radius of curvature that corresponds with the radius of curvature of the surface against which a continuous brush seal is to be formed during operation. The apparatus illustrated in Figs. 9A and 9B can be used to carry out such brush trimming operations. In the case of spillstrip segment manufacture, the radius of curvature associated with the cutting surface of the milling machine is set to the radius of curvature of the shroud band of the rotor. In the case of packing ring segment manufacture, the radius of curvature associated with the cutting surface of the milling machine is set to the radius of curvature.

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While the brush seal designs of the present invention have been illustrated in great detail in connection with improved spillstrips and packing rings, it is understood that such brush seal design can be applied to other subcomponents of elastic steam turbines. For example, the brush seal designs of the prevent invention can be used in "rotating" applications such as in the fabrication of: improved snout rings; improved bell seals; substitute seal for J strips and knurled strips (around nozzles); improved contact-type seals/packing (common on 1st Stage Of Reheat Section on large GE And Westinghouse turbines); improved upstream and downstream faces of diaphragms or wheels in order to create an actual seal, or to increase the surface roughness, create more turbulence in the flow, and effectively reduce flow area by reducing the area for laminar flow; particulate separator realized as by a ring of brushes (or rings in stages) at a location between the boiler and the turbine nozzle (which would serve to reduce steam path damage and reduces chance of particles jamming up retractable packing; seals as part of a particular separating pocket, in conjunction with coated tip seals; improved seals between/at bushings to prevent additional leakage or stem damage, and in conjunction with steam strainer; improved to seals at butt joints of packing/spills; and improved joints of shells/diaphragms/gland carriers.

Also, the brush seal design of the present invention can be used in "rotating" applications, such as in the fabrication of: substitutes for J-strips (on rotor or in diaphragm/blade rings); improved or replacement water/oil seals and/or hydrogen seals.

In Figs. 20 through 20A2, the brush seal technology of the present invention is used to realize horizontal joint seals. As shown in Fig. 20, the joint gap 55 formed between the top

and bottom casing portions 2A and 2B of the turbine is shown. Vapor leakage from the joint gap 55 is effectively sealed off by mounting brush-type segments 27 or 39 over the joint gap to form a continuous joint gap seal, as shown in Fig. 20. In Fig. 21, an alternative type of joint gap seal is realized using segments 27' or 39' having flexible metallic strips, rather than metallic bristles. The brush-type joint seal shown in Fig. 20 can be realized by mounting bristles within the body portion of mounting strip, in a manner similar to that shown in Fig. 5. The flexible-strip type joint gap seal segments can be made by mounting narrow strips of flexible material between a pair of thin plates. The body portion of the brush segments can be secured to interior wall surfaces of the casing, immediately above the joint gap 55, by welding, brazing, screws or other fasteners well known in the art.

As illustrated in Fig. 20A, the bristles, or flexible strips, of the brush-type joint gap sealing strip are mounted immediately above the joint gap of a turbine. As the interior pressure of the turbine increases during operation, the bristles (or flexible strip) become compressed and effectively seal off the joint gap. Figs. 20A1 and 20A2 illustrate the compression of the bristles and flexible strips, respectively, during operation of the brush seal of the present invention.

Various other modifications to the illustrative embodiment of the present invention will readily occur to persons with ordinary skill in the art. All such modifications and variations are deemed to be within the scope and spirit of the present invention as defined by the accompanying Claims to Invention.

WHAT IS CLAIMED IS:

1           1.    A packing ring segment for use in forming a labyrinth  
2 seal with a turbine shaft rotatably mounted within the casing of an  
3 axial flow elastic fluid turbine, wherein said axial flow elastic  
4 fluid turbine includes at least one stage having (i) at least one  
5 diaphragm stationarily mounted in said casting and having a  
6 plurality of steam directing nozzles, and (ii) a rotor fixedly  
7 attached to said turbine shaft and having a plurality of blades  
8 bounded by a shroud and disposed adjacent to said plurality of  
9 steam directing nozzles, said packing ring segment comprising:

10           a body portion for mounting within a portion of said diaphragm  
11 and having a longitudinal extent, a vertical extent, and a  
12 horizontal extent, and being particularly adapted for mounting in a  
13 portion of said diaphragm, said body portion having a groove formed  
14 along said vertical extent and extending along said longitudinal  
15 extent; and

16           a plurality of brush segments mounted within said groove of  
17 said packing ring segment, each said brush segment having a packet  
18 of bristles with said bristles having tip portions trimmed to  
19 terminate along a radius of curvature continuously extending along  
20 the longitudinal extent of said body portion so as to form a  
21 labyrinth seal with said turbine shaft, and

22           each said bristle being disposed in a plane substantially  
23 parallel to the principal plane of said rotor and extending in the  
24 direction of rotation of said turbine shaft.

1           2.    The packing ring segment of claim 1, wherein the bristles  
2 within each said bristle segment are bound between a pair of plates  
3 having an incised channel to receive the base end portion of said  
4 bristles.

1           3.    The packing ring segment of claim 1, wherein said plates  
2 of each said brush segment have a bottom edge surface with a radius  
3 of curvature adapted to conform to the radius of curvature of said  
4 groove formed in said vertical portion.

4. The packing ring segment of claim 1, wherein the bristles within each said brush segment are bound within a channel by a core strip securing said bristles within said channel.



1           5.    An elastic fluid turbine employing seals to minimize  
2   leakage between rotating and stationary components and utilizing  
3   the combination of a segmented seal ring supported by and at least  
4   partially contained in an annular groove formed in a stationary  
5   casing to permit motion of said seal ring between a large diameter  
6   position and a small diameter position corresponding respectively  
7   to large and small clearance of said seal ring with regard to the  
8   rotating shaft said seal ring groove being partially defined by a  
9   pair of opposing, spaced apart shoulders on said casing which form  
10   an opening of said groove extending radially into the clearance  
11   area between said casing and said rotating shaft, said turbine  
12   including at least one stage having (i) at least one diaphragm  
13   stationarily mounted in said casing and having a plurality of steam  
14   directing nozzles, and (ii) a rotor fixedly attached to said  
15   turbine shaft and having a plurality of blades bounded by a shroud  
16   and disposed adjacent to said plurality of steam directing nozzles,

17           each segment of said seal ring including an inner arcuate  
18   portion having seal teeth extending therefrom in the direction of  
19   and adjacent to said rotating shaft a radially outwardly facing  
20   arcuate surface on said seal ring segment which is located opposite  
21   to a radially inward facing arcuate surface on said casing for  
22   limiting said large clearance position by contact between said  
23   opposing surfaces, an outer ring portion disposed within said seal  
24   ring groove for both axial and radial movement therein and having a  
25   pair of shoulders extending axially in opposite directions for  
26   making radial contact respectively with said pair of spaced apart  
27   shoulders on said casing and thereby limiting said small clearance  
28   position;

29           a neck portion connected between said inner arcuate portion  
30   and said outer ring portion and extending between said casing  
31   shoulders, said neck portion having an axial thickness which is  
32   less than the distance between said opposing casing shoulders to  
33   thereby axially locate said seal ring segment against one of said  
34   casing shoulders and provide a contact pressure seal at the said  
35   side of said neck portion which is subject to lower turbine fluid  
36   pressure;

1 a radial positioning means comprising a spring against said  
2 ring segments to forcibly cause said to move to said large  
3 clearance position contacts, while working fluid which is flows  
4 into to the annular space between said casing and said ring  
5 segments will urge said segments toward said small clearance  
6 position, whereby at low speed and small turbine loads the spring  
7 forces wilt predominate, while at high flows and high working fluid  
8 pressure the pressure forces will predominate;

9 each of said ring segments comprising:

10 a body portion for mounting within a portion of said diaphragm  
11 and having a longitudinal extent, a vertical extent, and a  
12 horizontal extent, and being particularly adapted for mounting in a  
13 portion of said diaphragm, said body portion having a groove formed  
14 along said vertical extent and extending along said longitudinal  
15 extent; and

16 a plurality of brush segments mounted within said groove of  
17 said packing ring segment, each said brush segment having a packet  
18 of bristles with said bristles having tip portions trimmed to  
19 terminate along a radius of curvature continuously extending along  
20 the longitudinal extent of said body portion so as to form a  
21 labyrinth seal with said turbine shaft, and

22 each said bristle being disposed in a plane substantially  
23 parallel to the principal plane of said rotor and extending in the  
24 direction of rotation of said turbine shaft.

1 6. The fluid turbine seal arrangement as recited in claim 5,  
2 wherein said radial positioning means includes a flat spring  
3 interposed between said casing shoulders and an inner surface of  
4 said outer ring portion of said ring segment.

1           7.    The fluid turbine seal arrangement as recited in claim 5,  
2 wherein said inner arcuate portion of said seal ring segment  
3 comprises an inner ring portion connected at one side to said neck  
4 portion and includes said radially outward facing arcuate surface  
5 for limiting said large clearance position by said contact with  
6 said casing surface said inner ring portion having said seal teeth  
7 extending therefrom.

1           8.    The fluid turbine seal arrangement as recited in claim 7,  
2 where in said inner ring portion extends substantially from said  
3 neck portion axially in both directions such that said inner ring  
4 portion is substantially wider than said neck portion and has said  
5 seal teeth extending therefrom radially into said clearance area.

1           9.    The fluid turbine seal arrangement as recited in claim 8,  
2 wherein said radially outwardly facing arcuate surface in said seal  
3 ring segment which is located opposite to a radially inward facing  
4 arcuate surface of said casing comprises the radially outward  
5 facing surfaces of said inner ring portion.

1           10. A packing ring for use in forming a seal with a turbine  
2 shaft rotatably mounted within the casing of an axial flow elastic  
3 fluid turbine, wherein said axial flow elastic fluid turbine  
4 includes at least one stage having (i) at least one diaphragm  
5 stationarily mounted in said casting and having a plurality of  
6 steam directing nozzles, and (ii) a rotor fixedly attached to said  
7 turbine shaft and having a plurality of blades bounded by a shroud  
8 and disposed adjacent said plurality of steam directing nozzles,  
9 said packing ring comprising a plurality of packing ring segments,  
10 each said packing ring segment being mounted in a groove  
11 circumferentially in said diaphragm for forming a seal with said  
12 turbine shaft;

13           each said packing ring segment including  
14           a body portion for mounting within said circumferentially  
15 formed groove and having a longitudinal extent, a vertical extent,  
16 and a horizontal extent, and being particularly adapted for  
17 mounting in a portion of said diaphragm, said body portion having a  
18 brush mounting groove formed along said vertical extent and  
19 extending along said longitudinal extent; and

20           a plurality of brush segments mounted within said brush  
21 mounting groove of said packing ring segment, each said brush  
22 segment having a packet of bristles with said bristles having tip  
23 portions trimmed to terminate along a radius of curvature adapted  
24 to form a steam seal with said turbine shaft, and

25           each said bristle being disposed in a plane substantially  
26 parallel to the principal plane of said rotor and extending in the  
27 direction of rotation of said turbine shaft; and

28           each segment of said seal ring comprising:

29           an inner arcuate portion having seal teeth extending therefrom  
30 in the direction of and adjacent to said rotating shaft a radially  
31 outwardly facing arcuate surface on said seal ring segment which is  
32 located opposite to a radially inward facing arcuate surface on  
33 said casing for limiting said large clearance position by contact  
34 between said opposing surfaces, an outer ring portion disposed  
35 within said seal ring groove for both axial and radial movement  
36 therein and having a pair of shoulders extending axially in

1 opposite directions for making radial contact respectively with  
2 said pair of spaced apart shoulders on said casing and thereby  
3 limiting said small clearance position;

4 a neck portion connected between said inner arcuate portion  
5 and said outer ring portion and extending between said casing  
6 shoulders, said neck portion having an axial thickness which is  
7 less than the distance between said opposing casing shoulders to  
8 thereby axially locate said seal ring segment against one of said  
9 casing shoulders and provide a contact pressure seal at the said  
10 side of said neck portion which is subject to lower turbine fluid  
11 pressure; and

12 a radial positioning means comprising a spring against said  
13 ring segments to forcibly cause said to move to said large  
14 clearance position contacts, while working fluid which is flows  
15 into to the annular space between said casing and said ring  
16 segments will urge said segments toward said small clearance  
17 position, whereby at low speed and small turbine loads the spring  
18 forces wilt predominate, while at high flows and high working fluid  
19 pressure the pressure forces will predominate;

1 11. The packing ring of claim 10, wherein the bristles within  
2 each said bristle segment are bound between a pair of plates having  
3 an incised channel to receive the base end portion of said  
4 bristles.

1 12. The packing ring of claim 11, wherein said plates of each  
2 said brush segment have a bottom edge surface with a radius of  
3 curvature adapted to conform to the radius of curvature of said  
4 groove formed in said vertical portion.

1 13. The packing ring of claim 10, wherein the bristles within  
2 each said brush segment are bound within a channel by a core strip  
3 securing said bristles within said channel.

14. The packing ring of claim 10, wherein said channel has a bottom edge surface with a radius of curvature adapted to conform to the radius of curvature of said groove formed in said vertical portion.

15. An axial flow elastic fluid turbine comprising:  
an outer casting;  
a turbine shaft rotatably supported in said outer casting; and  
a plurality of turbine stages installed along said turbine shaft and contained within said outer casting, each said turbine stage including

a diaphragm stationarily mounted in a recess formed in said turbine casting and having a plurality of steam directing nozzles,

a rotor fixedly attached to said turbine shaft and having a plurality of blades bounded by a shroud band and being disposed adjacent said plurality of steam directing nozzles, and

a packing ring consisting of an arrangement of packing ring segments mounted in a circumferentially extending groove formed in said diaphragm and providing a continuously extending seal with said turbine,

each said packing ring segment including

a body portion for mounting within said circumferentially formed groove and having a longitudinal extent, a vertical extent, and a horizontal extent, and being particularly adapted for mounting in a portion of said diaphragm, said body portion having a brush mounting groove formed along said vertical extent and extending along said longitudinal extent; and

a plurality of brush segments mounted within said brush mounting groove of said packing ring segment, each said brush segment having a packet of bristles with said bristles having tip portions trimmed to terminate along a radius of curvature adapted to form a steam seal with said turbine shaft, and

each said bristle being disposed in a plane substantially parallel to the principal plane of said rotor and extending in the direction of rotation of said turbine shaft.

1           16. The axial flow elastic fluid turbine of claim 15, wherein  
2 the bristles within each said bristle segment are bound between a  
3 pair of plates having an incised channel to receive the base end  
4 portion of said bristles.

1           17. The axial flow elastic fluid turbine of claim 15, wherein  
2 said plates of each said brush segment have a bottom edge surface  
3 with a radius of curvature adapted to conform to the radius of  
4 curvature of said groove formed in said vertical portion.

1           18. The axial flow elastic fluid turbine of claim 15, wherein  
2 the bristles within each said brush segment are bound within a  
3 channel by a core strip securing said bristles within said channel.

1           19. The axial flow elastic fluid turbine of claim 18, wherein  
2 said channel has a bottom edge surface with a radius of curvature  
3 adapted to conform to the radius of curvature of said groove formed  
4 in said vertical portion.

1           20. The axial flow elastic fluid turbine as recited in claim  
2 15, wherein said radial positioning means includes a flat spring  
3 interposed between said casing shoulders and an inner surface of  
4 said outer ring portion of said ring segment.

1           21. The axial flow elastic fluid turbine as recited in claim  
2 15, wherein said inner arcuate portion of said seal ring segment  
3 comprises an inner ring portion connected at one side to said neck  
4 portion and includes said radially outward facing arcuate surface  
5 for limiting said large clearance position by said contact with  
6 said casing surface said inner ring portion having said seal teeth  
7 extending therefrom.

1           22. The axial flow elastic fluid turbine as recited in claim  
2 21, where in said inner ring portion extends substantially from  
3 said neck portion axially in both directions such that said inner  
4 ring portion is substantially wider than said neck portion and has  
5 said seal teeth extending therefrom radially into said clearance  
6 area.

1           23. The axial flow elastic fluid turbine as recited in claim  
2 22, wherein said radially outwardly facing arcuate surface in said  
3 seal ring segment which is located opposite to a radially inward  
4 facing arcuate surface of said casing comprises the radially  
5 outward facing surfaces of said inner ring portion.



1           24. A packing ring segment for use in forming a labyrinth  
2 seal with a turbine shaft rotatably mounted within the casing of  
3 an axial flow elastic fluid turbine, wherein said axial flow  
4 elastic fluid turbine includes at least one stage having (i) at  
5 least one diaphragm stationarily mounted in said casing and having  
6 a plurality of steam directing nozzles, and (ii) a rotor fixedly  
7 attached to said turbine shaft for rotation about an axis of  
8 rotation and having a plurality of blades bounded by a shroud band  
9 disposed within a principal plane of said rotor and adjacent said  
10 plurality of steam directing nozzles, said packing ring segment  
11 comprising:

12           a body portion for mounting within a portion of said diaphragm  
13 and having a longitudinal extent, a vertical extent, and a  
14 horizontal extent, and being particularly adapted for mounting in a  
15 portion of said diaphragm, said body portion having a groove formed  
16 along said vertical extent and extending along said longitudinal  
17 extent; and

18           a plurality of brush segments mounted within said groove of  
19 said packing ring segment, each said brush segment having a packet  
20 of bristles with said bristles having tip portions trimmed to  
21 terminate along a radius of curvature continuously extending along  
22 the longitudinal extent of said body portion so as to form a  
23 labyrinth seal with said turbine shaft, and

24           each said bristle being disposed at an acute angle with  
25 respect to the principal plane of said rotor.

1           25. The packing ring segment of claim 20, wherein the  
2 bristles within each said bristle segment are bound between a pair  
3 of plates having an incised channel to receive the base end portion  
4 of said bristles.

1           26. The packing ring segment of claim 21, wherein said plates  
2 of each said brush segment have a bottom edge surface with a radius  
3 of curvature adapted to conform to the radius of curvature of said  
4 groove formed in said vertical portion.

1           27. The packing ring segment of claim 20, wherein the  
2 bristles within each said brush segment are bound within a channel  
3 by a core strip securing said bristles within said channel.

1           28. A turbine or compressor comprising at least one packing  
2 ring segment as defined by claim 20.

# ABSTRACT

Disclosed are improved spillstrip and packing rings, and packing ring segments employing novel brush elements for use in forming seals in axial flow elastic fluid machines, and compressor and turbine machines containing the same, .

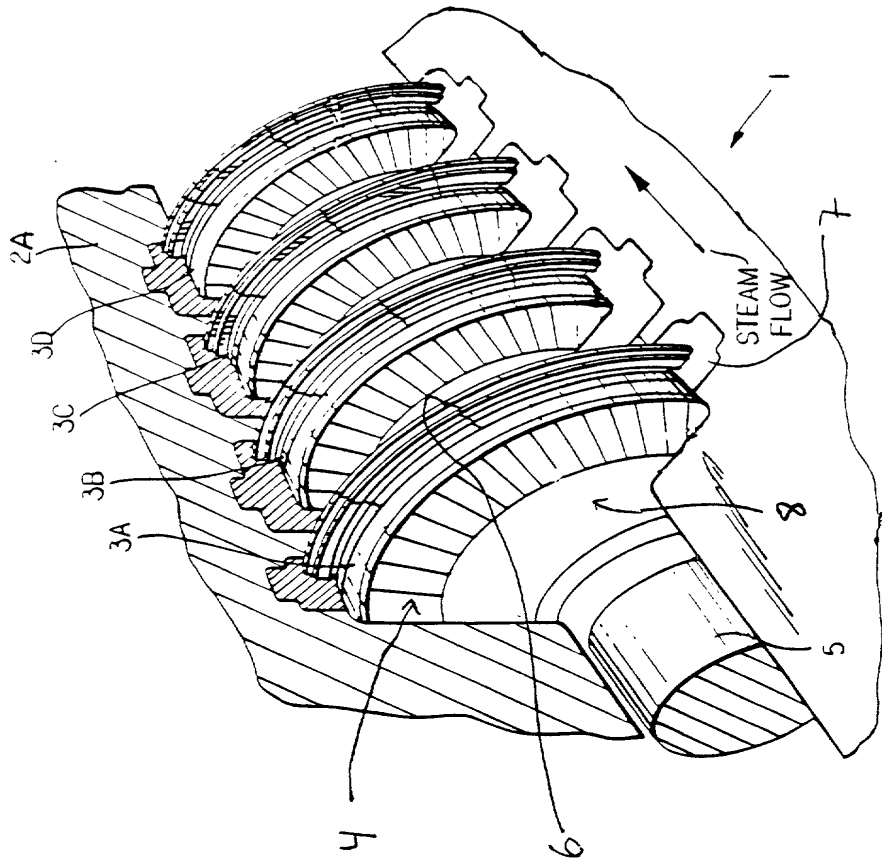
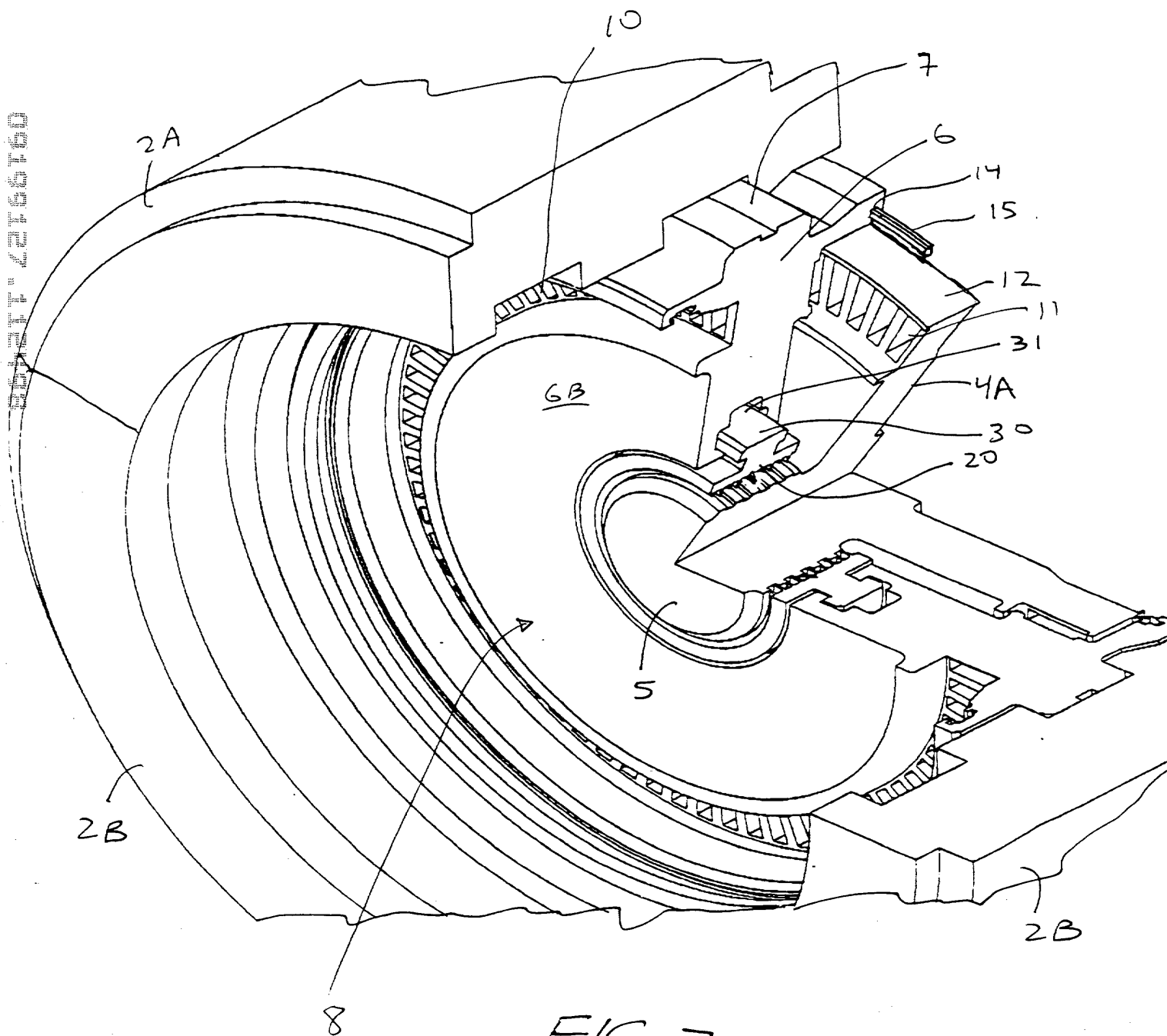
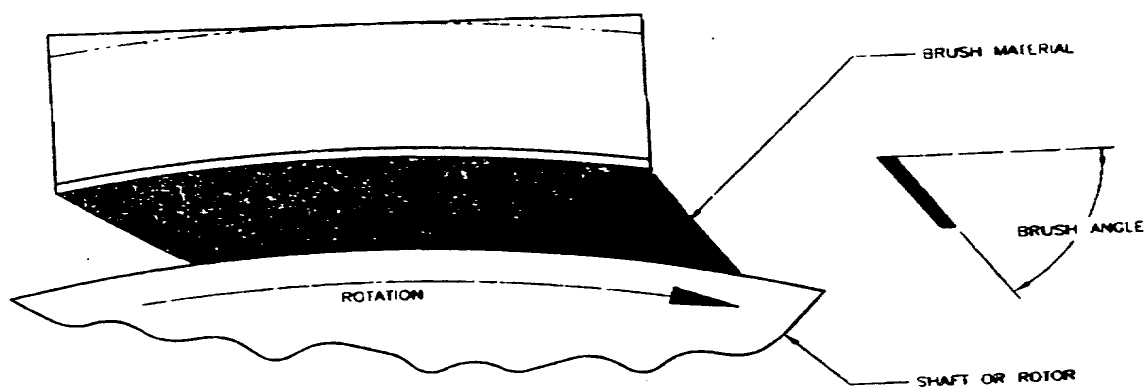
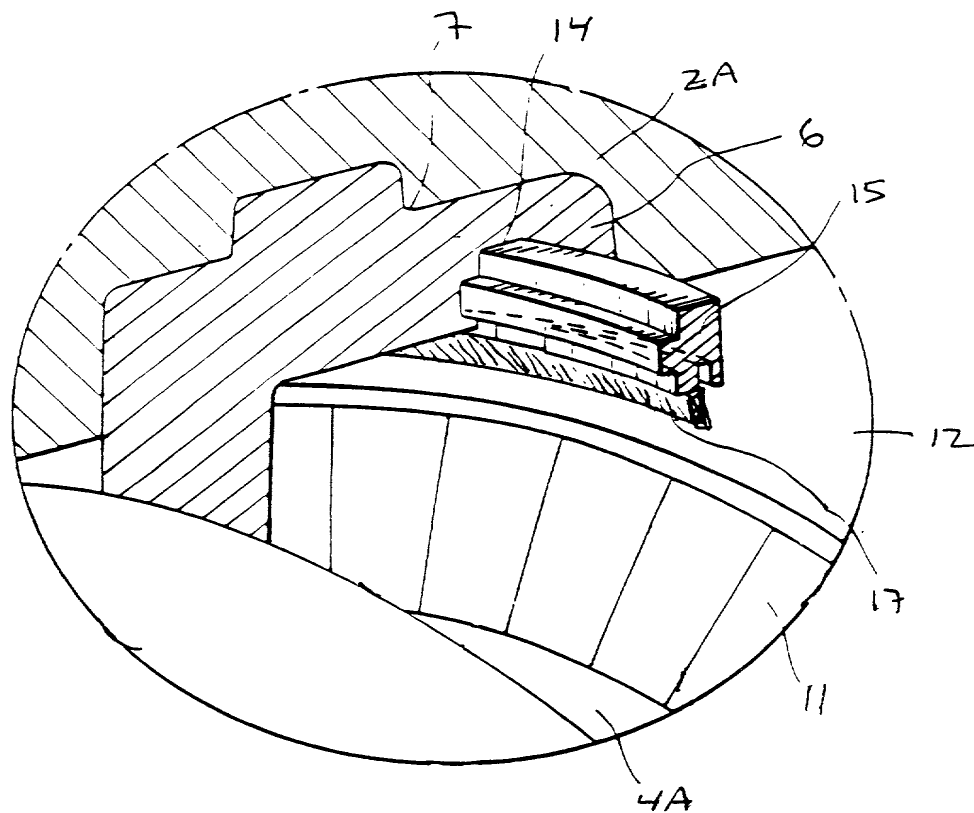


FIG 1

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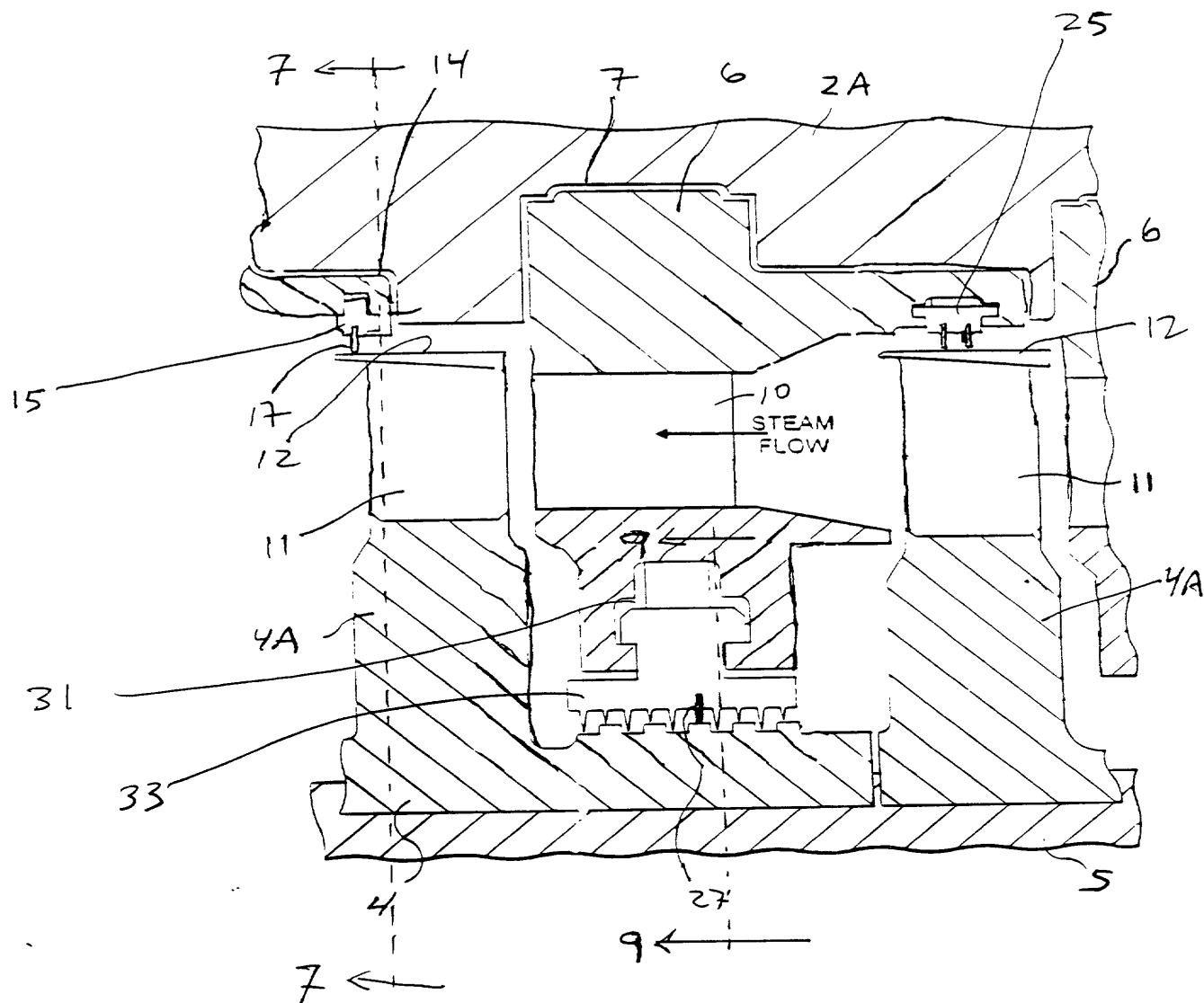
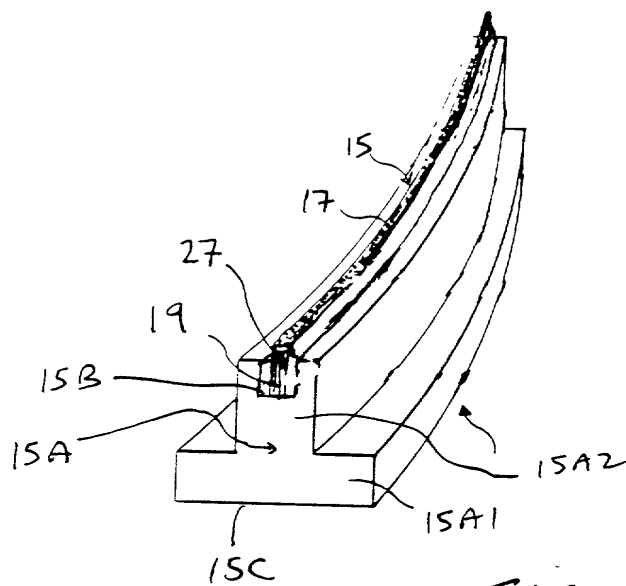


FIG. 3



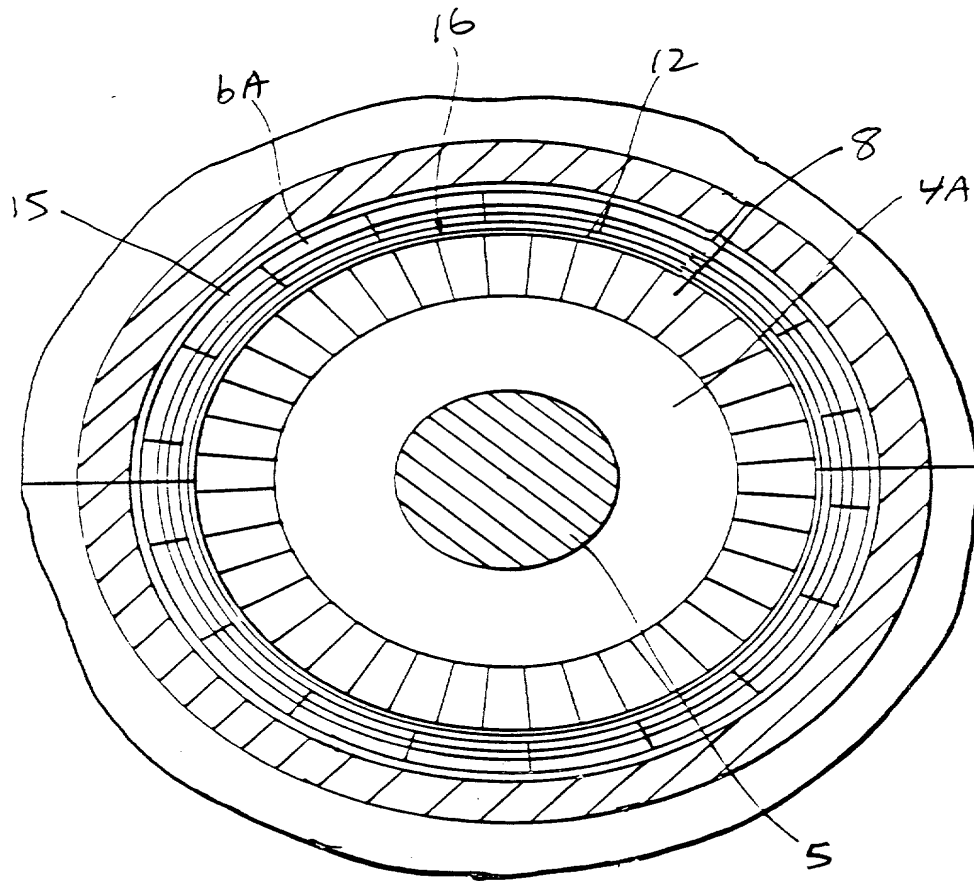
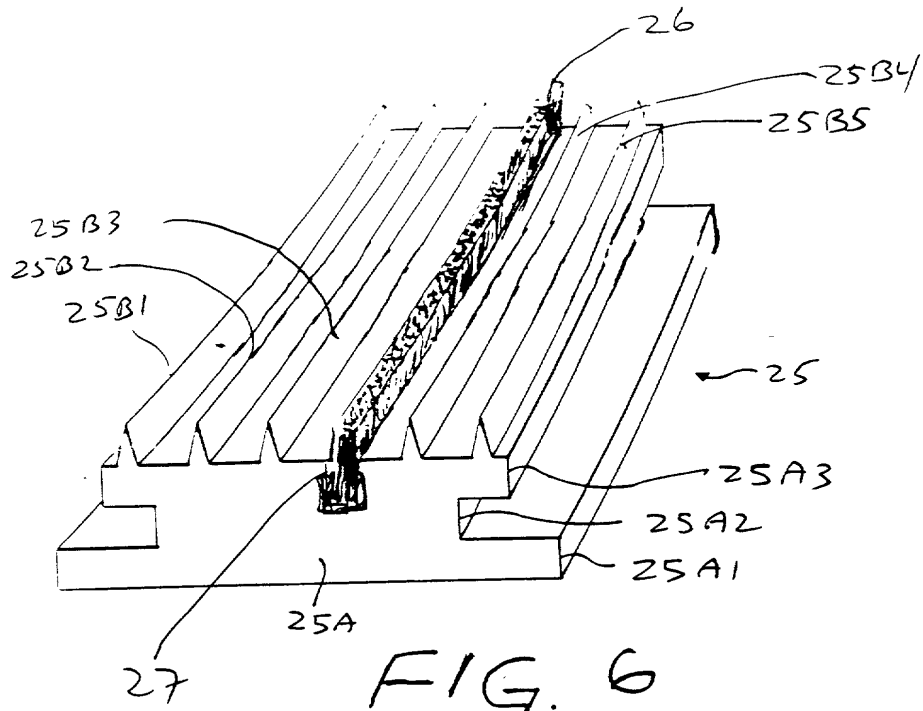


**REPORT**

By  
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MILNOR  
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JOHN  
W. BROWN



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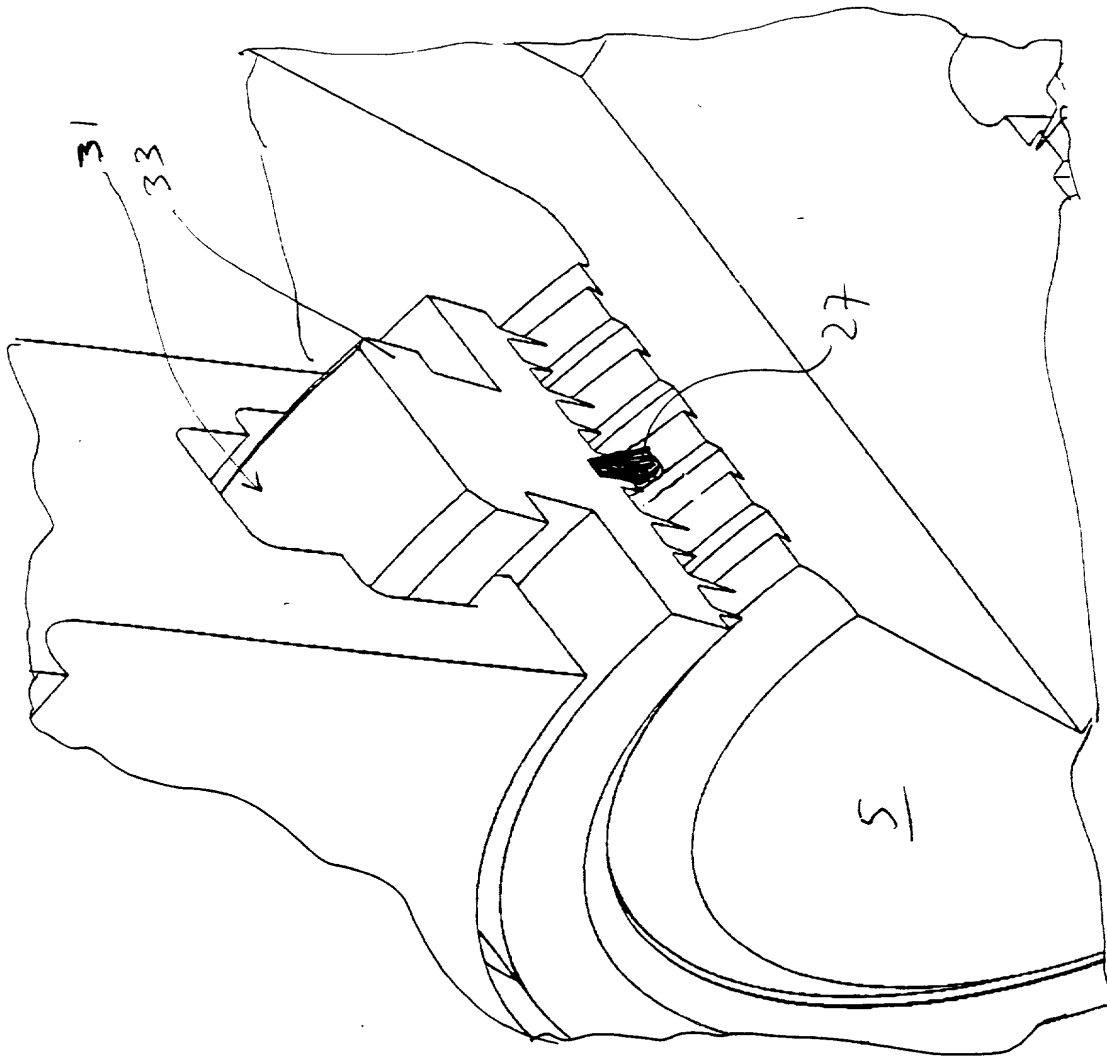


FIG. 8

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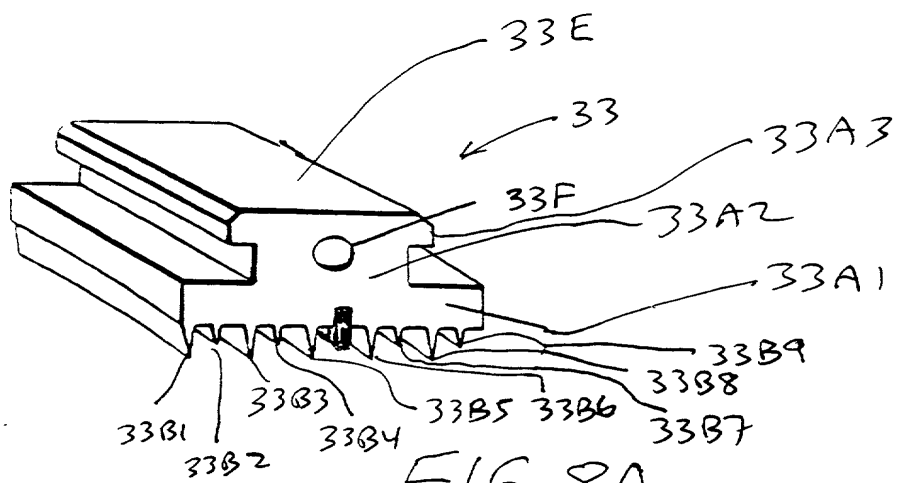


FIG. 8A

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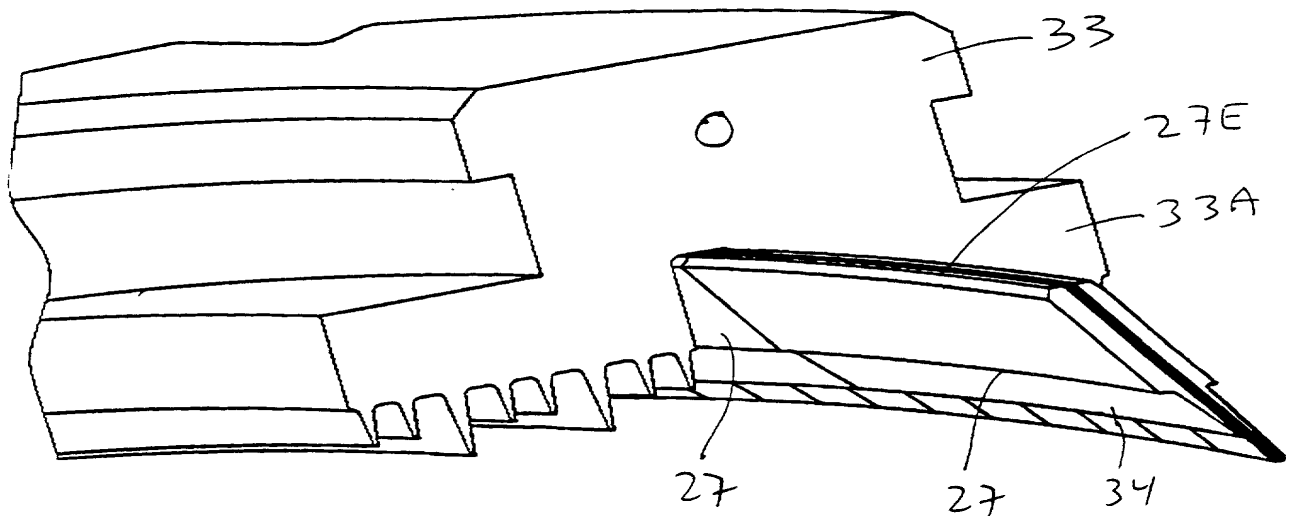


FIG. 8B

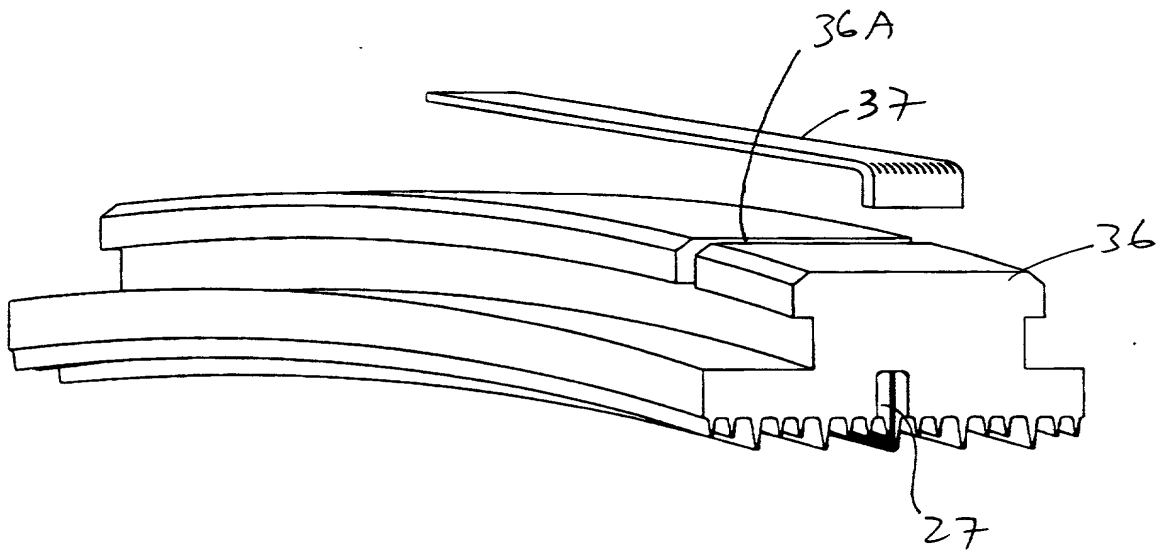


FIG. 8C

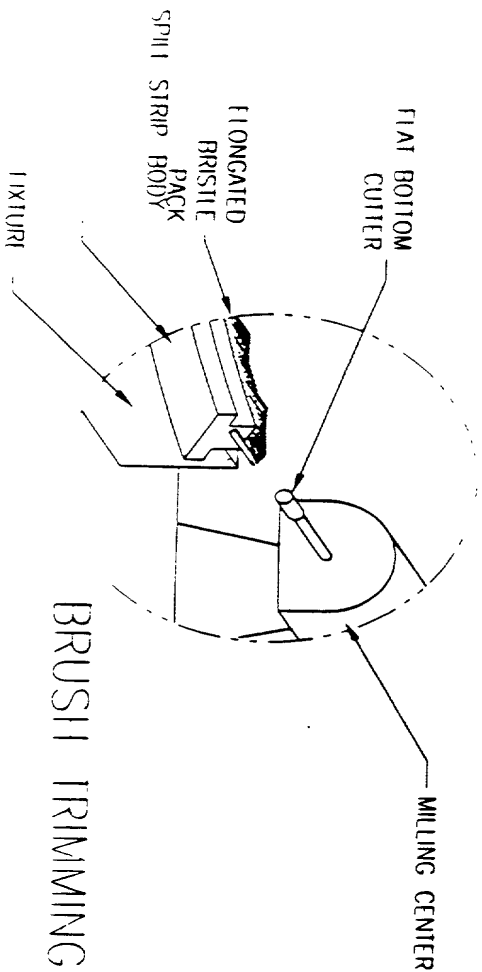
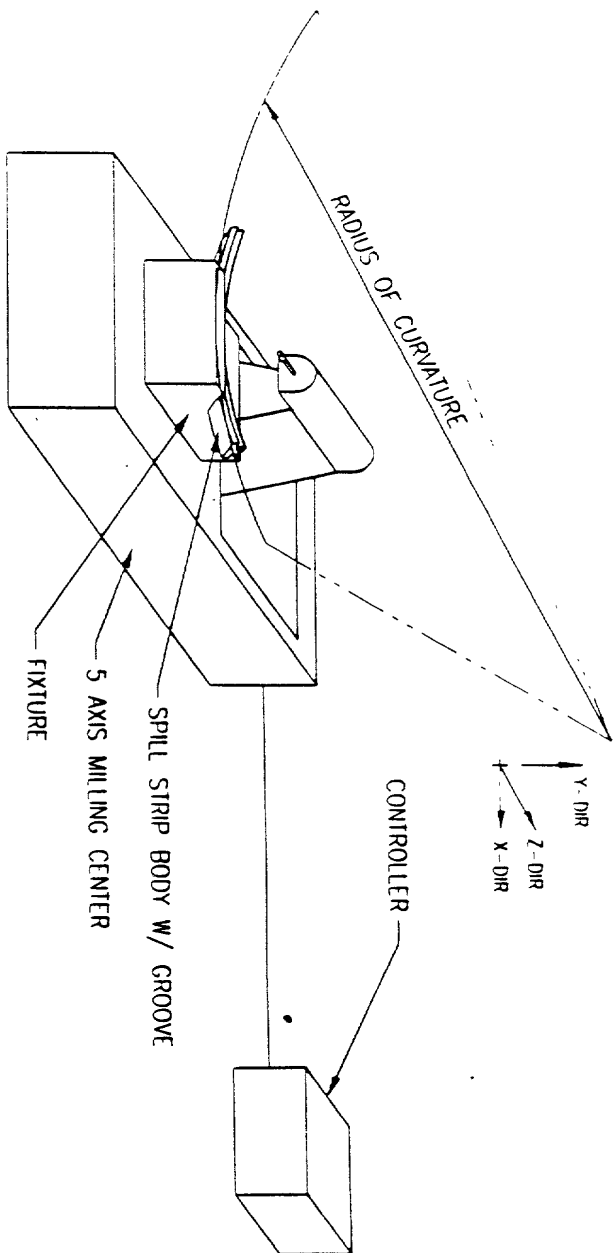
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# SETUP FOR MACHINING SPILL STRIP BODIES

FIG 9A



## BRUSH TRIMMING OPERATION

FIG 9B

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# Manufacturing Flowchart

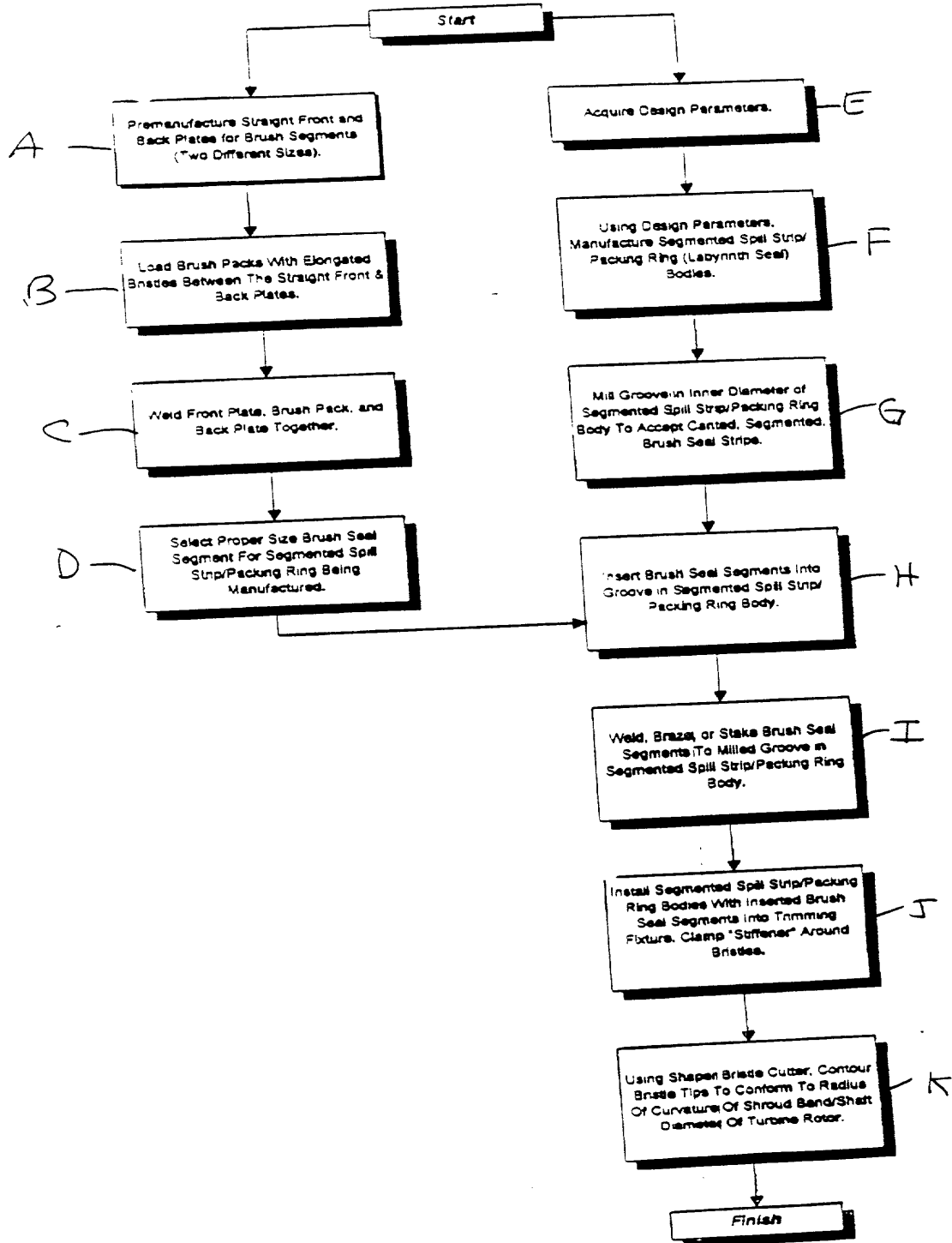


FIG 10

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## Segmented Brush Seal Drop Study

Based On Curved Segments

&gt;&gt;&gt;

Radius of Curvature for Segment 15.530

	Segment Length (in inches)										
	1/8	1/2	3/4	1	1 1/4	1 1/2	2	2 1/2	3	3 1/2	4
20	0.0006	0.0011	0.0025	0.0044	0.0069	0.0100	0.0178	0.0279	0.0404	0.0552	0.0725
21	0.0005	0.0010	0.0022	0.0038	0.0060	0.0087	0.0154	0.0242	0.0349	0.0478	0.0627
22	0.0005	0.0008	0.0019	0.0033	0.0052	0.0074	0.0133	0.0208	0.0300	0.0410	0.0538
23	0.0004	0.0007	0.0016	0.0028	0.0044	0.0063	0.0113	0.0177	0.0255	0.0348	0.0457
24	0.0003	0.0006	0.0013	0.0024	0.0037	0.0053	0.0094	0.0148	0.0214	0.0292	0.0383
25	0.0003	0.0006	0.0011	0.0019	0.0030	0.0044	0.0078	0.0122	0.0178	0.0240	0.0315
26	0.0002	0.0004	0.0009	0.0016	0.0024	0.0035	0.0062	0.0098	0.0141	0.0192	0.0252
27	0.0002	0.0003	0.0007	0.0012	0.0019	0.0027	0.0048	0.0075	0.0108	0.0148	0.0194
28	0.0001	0.0002	0.0005	0.0009	0.0014	0.0019	0.0035	0.0054	0.0078	0.0107	0.0140
29	0.0001	0.0001	0.0003	0.0006	0.0009	0.0013	0.0022	0.0035	0.0050	0.0069	0.0090
30	0.0000	0.0001	0.0002	0.0003	0.0004	0.0006	0.0011	0.0017	0.0024	0.0033	0.0044
31	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
32	0.0000	0.0001	0.0001	0.0003	0.0004	0.0006	0.0010	0.0016	0.0023	0.0031	0.0041
33	0.0001	0.0001	0.0003	0.0005	0.0008	0.0011	0.0020	0.0031	0.0044	0.0060	0.0079
34	0.0001	0.0002	0.0004	0.0007	0.0011	0.0016	0.0029	0.0045	0.0064	0.0088	0.0115
35	0.0001	0.0002	0.0005	0.0009	0.0014	0.0021	0.0037	0.0058	0.0083	0.0114	0.0149

Packing Ring Slot Diameter (in inches)

Drop (Δ Radius of Curvature) (in inches)

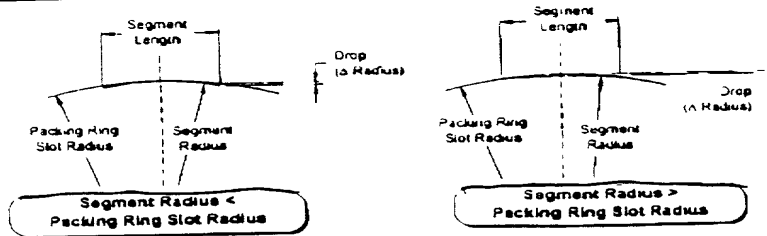


FIG. 11A

## Segmented Brush Seal Drop Study

Based On Curved Segments

&gt;&gt;&gt;

Radius of Curvature for Segment 11.500

	Segment Length (in inches)										
	1/8	1/2	3/4	1	1 1/4	1 1/2	2	2 1/2	3	3 1/2	4
29	0.0002	0.0004	0.0009	0.0016	0.0026	0.0037	0.0066	0.0103	0.0149	0.0204	0.0268
21	0.0001	0.0003	0.0006	0.0010	0.0016	0.0023	0.0042	0.0065	0.0094	0.0129	0.0170
22	0.0001	0.0001	0.0003	0.0005	0.0008	0.0011	0.0020	0.0031	0.0045	0.0062	0.0081
23	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
24	0.0001	0.0001	0.0003	0.0005	0.0007	0.0010	0.0018	0.0029	0.0041	0.0056	0.0074
25	0.0001	0.0002	0.0005	0.0009	0.0014	0.0020	0.0035	0.0055	0.0079	0.0108	0.0142
26	0.0002	0.0003	0.0007	0.0013	0.0020	0.0028	0.0050	0.0079	0.0114	0.0156	0.0205
27	0.0002	0.0004	0.0009	0.0016	0.0025	0.0036	0.0065	0.0101	0.0147	0.0200	0.0263
28	0.0003	0.0006	0.0011	0.0019	0.0030	0.0044	0.0078	0.0122	0.0177	0.0241	0.0317
29	0.0003	0.0006	0.0013	0.0023	0.0035	0.0051	0.0090	0.0142	0.0205	0.0279	0.0367
30	0.0004	0.0008	0.0014	0.0025	0.0040	0.0057	0.0102	0.0160	0.0231	0.0315	0.0413
31	0.0004	0.0007	0.0016	0.0028	0.0044	0.0063	0.0113	0.0177	0.0255	0.0348	0.0457
32	0.0004	0.0008	0.0017	0.0031	0.0048	0.0069	0.0123	0.0192	0.0278	0.0379	0.0498
33	0.0005	0.0008	0.0019	0.0033	0.0052	0.0074	0.0132	0.0207	0.0299	0.0409	0.0536
34	0.0006	0.0009	0.0020	0.0035	0.0055	0.0078	0.0141	0.0221	0.0319	0.0436	0.0572
35	0.0005	0.0009	0.0021	0.0037	0.0058	0.0084	0.0150	0.0234	0.0338	0.0462	0.0606

Packing Ring Slot Diameter (in inches)

Drop (Δ Radius of Curvature) (in inches)

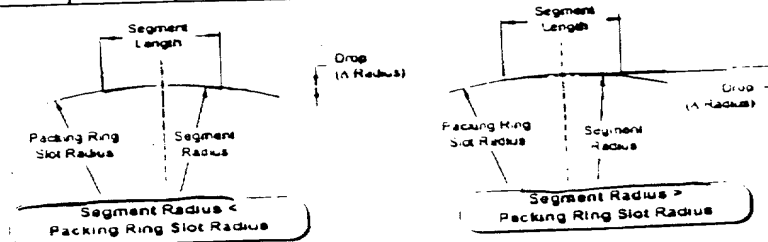
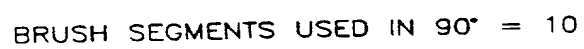


FIG. 11B

1. The first part of the book is a general introduction to the study of the history of the world, and is divided into two main sections: the first section deals with the history of the world from the beginning of time to the present, and the second section deals with the history of the world from the present to the future.



BRUSH SEGMENTS USED IN 90° = 13

FIG. 12B

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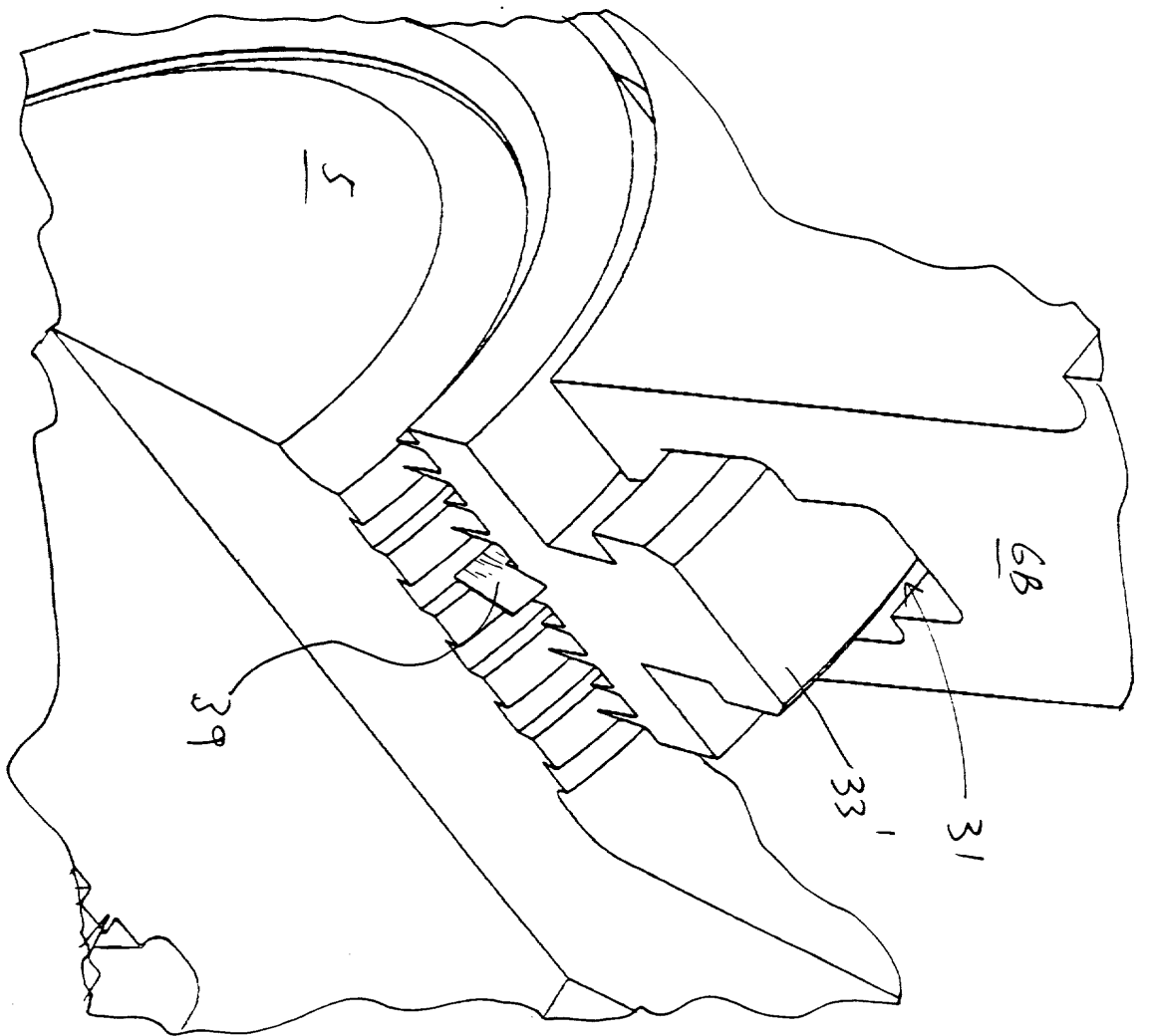


FIG. 13

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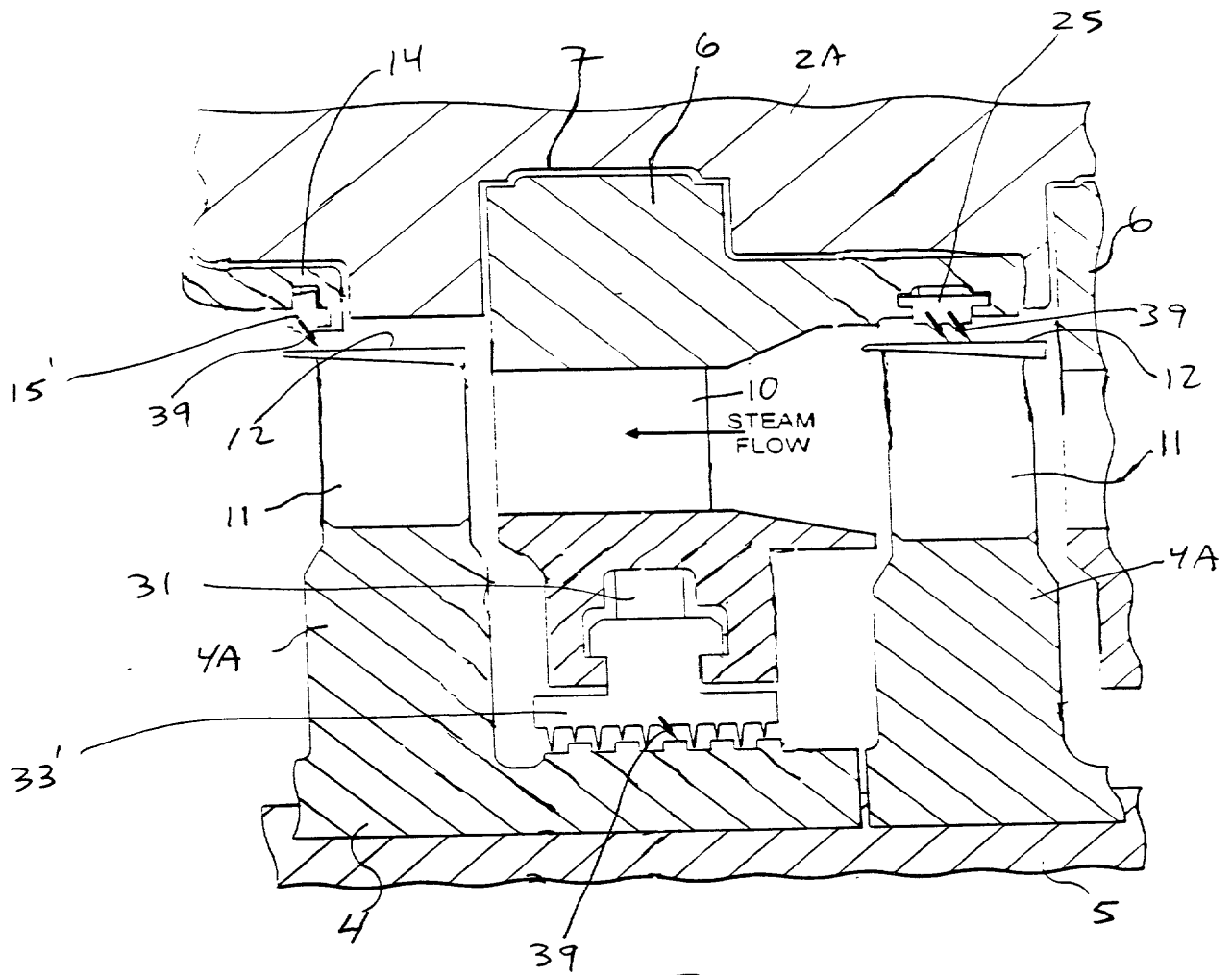


FIG. 14

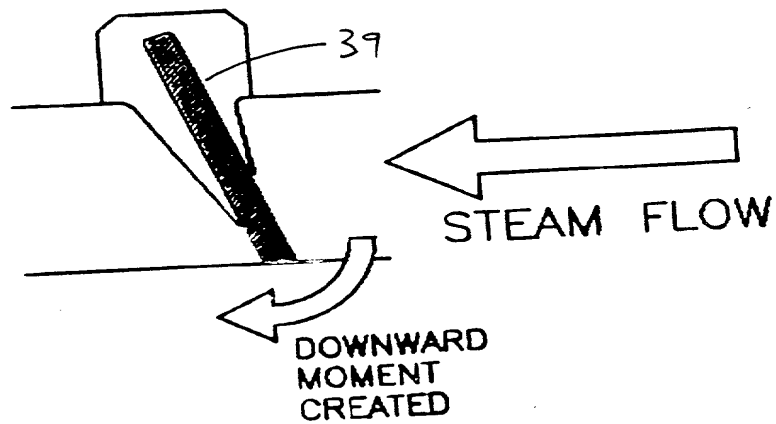
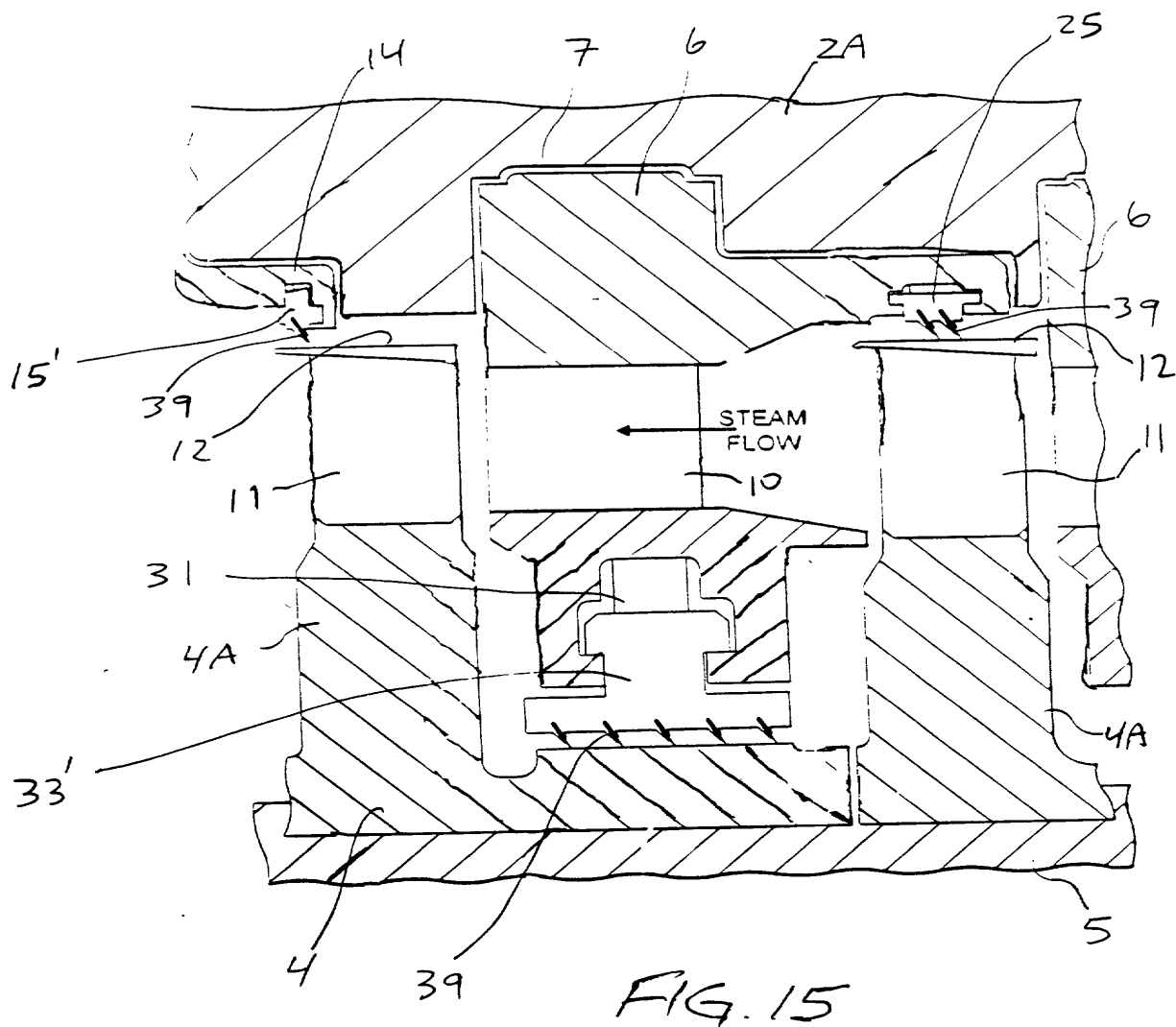


FIG. 14A

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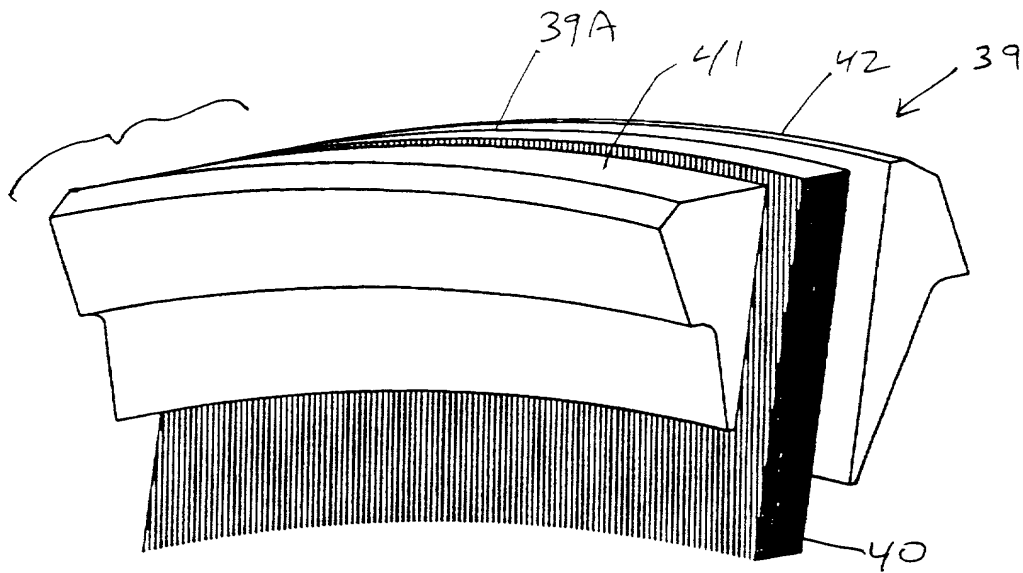


FIG. 16

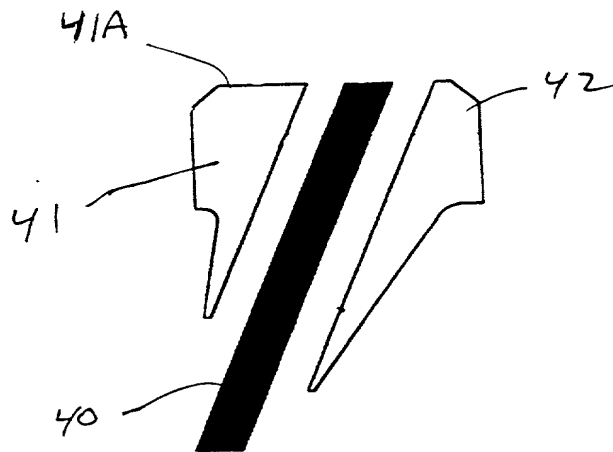


FIG. 16A

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# Manufacturing Flowchart

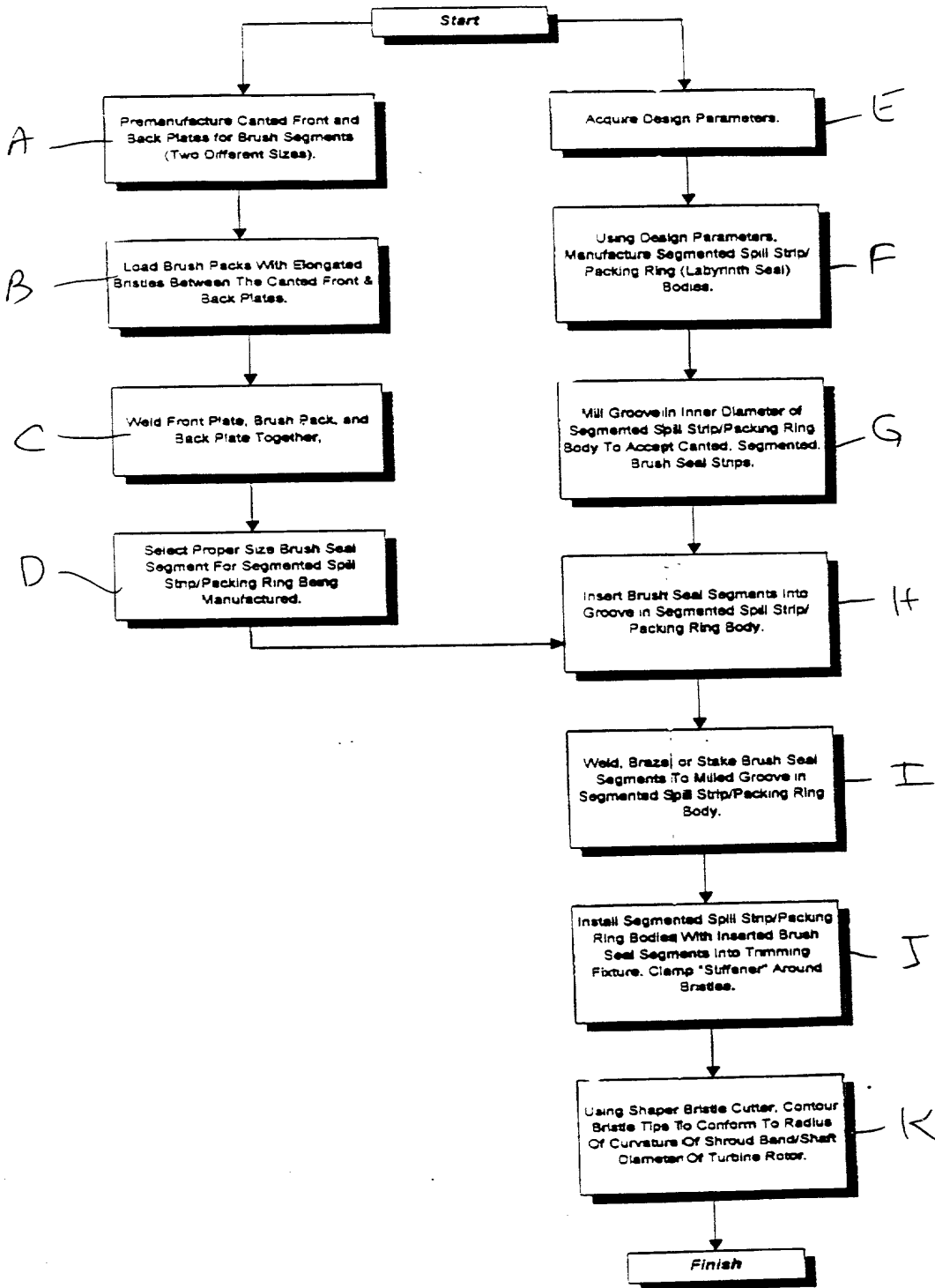
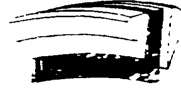


FIG. 17

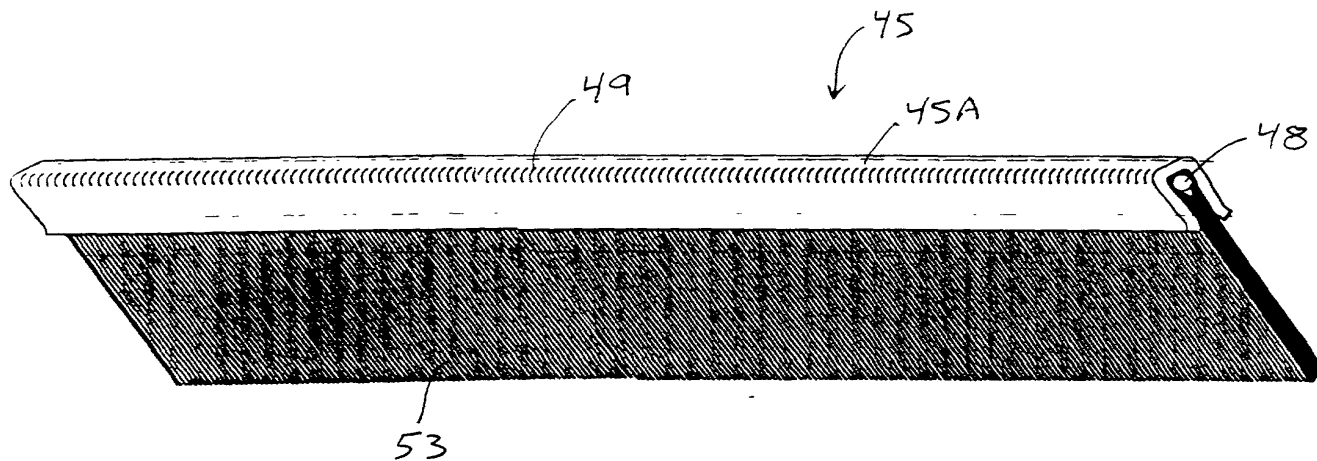


FIG. 18

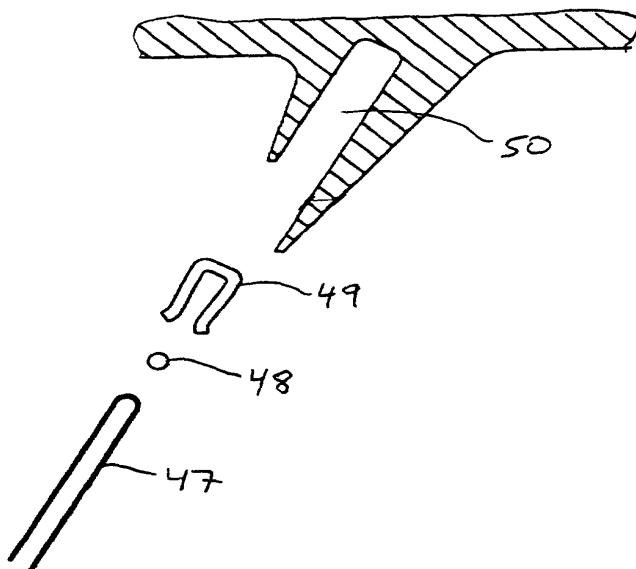


FIG 18A

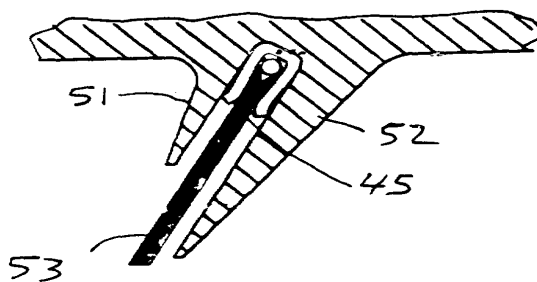


FIG 18B



# Manufacturing Flowchart

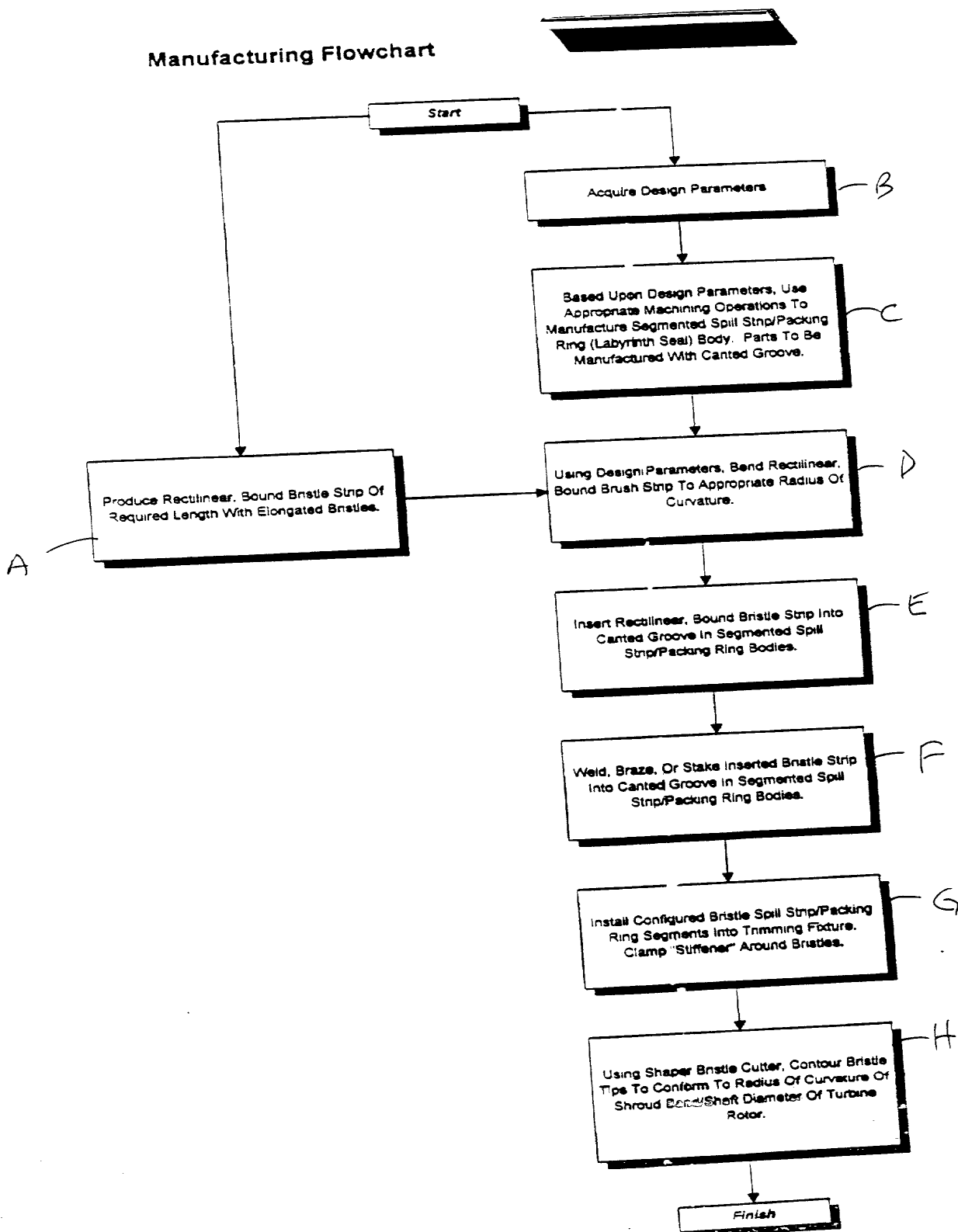


FIG. 19

# BRUSH SEALING HORIZONTAL JOINT OF A TURBINE SHELL

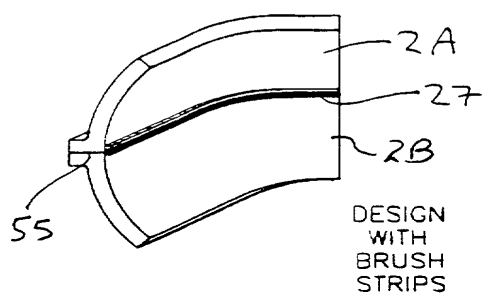


FIG. 20

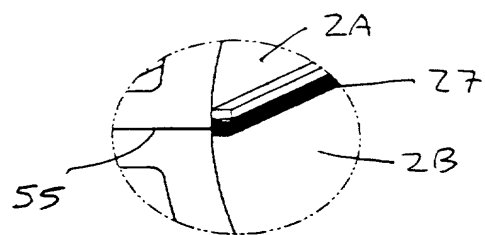


FIG. 20A

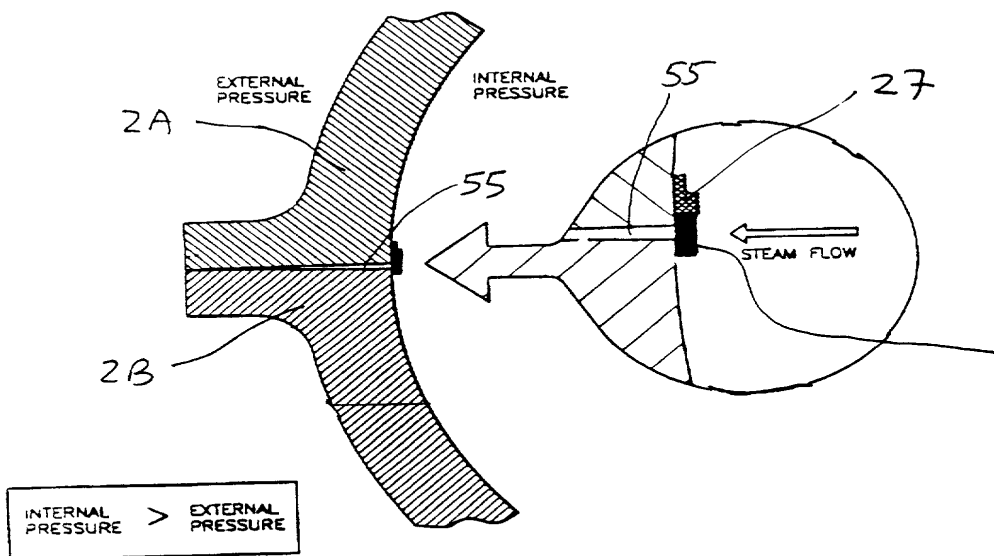


FIG. 20B

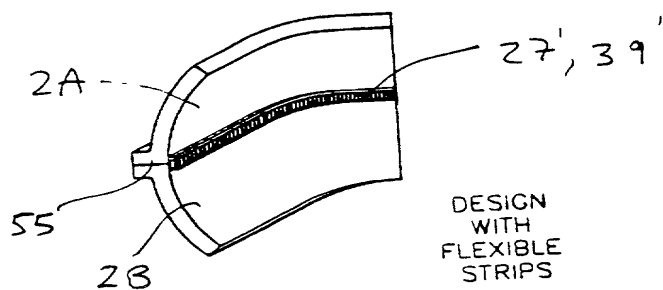


FIG. 21

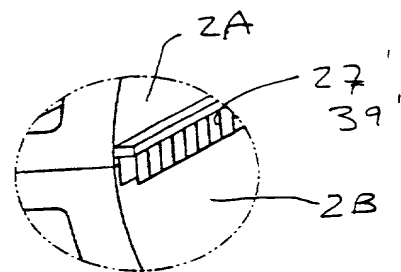


FIG. 21A

**DECLARATION AND POWER OF ATTORNEY FOR PATENT APPLICATION**

As a below named inventor, I hereby declare that:

My residence or post office address, and citizenship, are as stated below next to my name and signature.

I believe I am an original, first and sole inventor, if only one name is shown below, or an original, first, and joint inventor, if more than one name is shown below, of the subject matter which is claimed and for which a patent is sought on the invention entitled "**Improved Brush-Seal Designs and Elastic Fluid Turbines**", the specification of which was filed on **July 11, 1997** and accorded serial number **08/891,526**.

I hereby state that I have reviewed and understand the contents of the above-identified specification, including the claims, as amended by any amendment referred to above.

I acknowledge the duty to disclose information which is material to the examination of this application in accordance with Title 37, Code of Federal Regulations, §1.56(a).

**POWER OF ATTORNEY:** As a named inventor, I hereby appoint the following attorneys to prosecute this application and transact all business in the U.S. Patent and Trademark Office connected therewith, Hopgood, Calimafde, Kalil & Judlowe, LLP, a firm comprising Roy C. Hopgood, Reg. No. 15,245; John M. Calimafde, Reg. No. 16,895; Eugene J. Kalil, Reg. No. 16,686; Marvin N. Gordon, Reg. No. 23,094; Stephen B. Judlowe, Reg. No. 21,049; Francis J. Murphy, Reg. No. 24,537; Dennis J. Mondolino, Reg. No. 27,148; William G. Todd, Reg. No. 28,480; Ira B. Winkler, Reg. No. 29,223; James M. Bollinger, Reg. No. 32,555; Porter F. Fleming, Reg. No. 31,759; and Brian P. Murphy, Reg. No. 34,986; and associated therewith Bradley N. Ruben, Reg. No. 32,058.

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I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code and that such willful false statements may jeopardize the validity of the application or any patent issued thereon.

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SECRET 00493127